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**Laser-produced porous surface**

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## ABSTRACT

A method of fabricating a porous or partially porous three-dimensional metal article for use as a tissue ingrowth surface on a prosthesis. The porous article is formed using direct laser remelting in a cross section of a layer of metallic powder on a build platform without fusing thereto. The power, speed, spot size and beam overlap of the scanning laser is coordinated so that a predetermined porosity of the metallic powder can be achieved. Laser factors also vary depending from the thickness of the powder layer, type of metallic powder and size and size distribution of the powder particles. Successive depositing and remelting of individual layers are repeated until the article is fully formed by a layer-by-layer fashion. In an additional embodiment, a first layer of metallic powder may be deposited on a solid base or core and fused thereto.

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## LASER-PRODUCED POROUS SURFACE

### BACKGROUND OF THE INVENTION

The present invention relates to a porous surface and a method for forming the same, which uses a directed energy beam to selectively remelt a powder to produce a part. In particular, this invention relates to a computer-aided laser apparatus, which sequentially remelts a plurality of powder layers to build the designed part in a layer-by-layer fashion. The present application is particularly directed toward a method of forming a porous and partially porous metallic structure.

### DESCRIPTION OF THE RELEVANT ART

The field of free-form fabrication has seen many important recent advances in the fabrication of articles directly from computer controlled databases. These advances, many of which are in the field of rapid prototyping of articles such as prototype parts and mold dies, have greatly reduced the time and expense required to fabricate articles, particularly in contrast to conventional machining processes in which a block of material, such as a metal, is machined according to engineering drawings.

One example of a modern rapid prototyping technology is the selective laser sintering process practiced by systems available from DTM Corporation of Austin, Texas. According to this technology, articles are produced in layer-wise fashion from a laser-fusible powder that is dispensed one layer at a time. The powder is fused, remelted or sintered, by the application of laser energy that is directed in raster-scan fashion to portions of the powder layer corresponding to a cross section of the article. After the fusing of the powder in each layer, an additional layer of powder is dispensed, and

the process repeated, with fused portions or lateral layers fusing so as to fuse portions of previous laid layers until the article is complete. Detailed descriptions of the selective laser sintering technology may be found in U.S. Patent No. 4,863,538, U.S. Patent No. 5,017,753, U.S. Patent No. 5,076,869 and U.S. Patent No. 4,944,817, all assigned to Board of Regents, the University of Texas. Quasi-porous structures have also been developed in the form of regular and irregular lattice arrangements in which individual elements (column and connecting cross-members) are constructed singularly from a pre-defined computer-aided design model of the external geometry and lattice structure. The selective laser remelting and sintering technologies have enabled the direct manufacture of solid or dense three-dimensional articles of high resolution and dimensional accuracy from a variety of materials including wax, metal powders with binders, polycarbonate, nylon, other plastics and composite materials, such as polymer-coated metals and ceramics.

The field of the rapid prototyping of parts has, in recent years, made large improvements in broadening high strain, high density, parts for use in the design and pilot production of many useful articles, including metal parts. These advances have permitted the selective laser remelting and sintering processes to now also be used in fabricating prototype tooling for injection molding, with expected tool life in excess of ten thousand mold cycles. The technologies have also been applied to the direct fabrication of articles, such as molds, from metal powders without a binder. Examples of metal powder reportedly used in such direct fabrication include two-phase metal powders of the copper-tin, copper-solder (the solder being 70% lead and 30% tin), and bronze-nickel systems. The metal articles formed in these ways have been quite dense, for

example, having densities of up to 70% to 80% of fully dense (prior to any infiltration). Prior applications of this technology have strived to increase the density of the metal structures formed by the remelting or sintering processes. The field of rapid prototyping of parts has focused on providing high strength, high density, parts for use and design in production of many useful articles, including metal parts.

However, while the field of rapid prototyping has focused on increasing density of such three-dimensional structures, the field has not focused its attention on reducing the density of three-dimensional structures. Consequently, applications where porous and partially porous metallic structures, and more particularly metal porous structures with interconnected porosity, are advantageous for use have been ignored. It is an object of this invention to use a laser and powder metal to form pores in structures in which pores are irregular in size and have a controlled total porosity. It is a further object to produce porous tissue in growth surfaces with interconnected porosity with uniform pores and porosity.

#### SUMMARY OF THE INVENTION

The present invention relates to a method for producing a three-dimensional porous structure particularly for use with tissue ingrowth. In one embodiment of the present invention, a layer of metallic powder is deposited onto a substrate or a build platform. Preferred metals for the powder include titanium, titanium alloys, stainless steel, cobalt chrome alloys, tantalum or niobium. A laser beam with predetermined settings scans the powder layer causing the powder to preferentially remelt and consequently solidify with a decreased density, resulting from an increase in porosity as compared to a solid metal. The range of the laser's power may be between 5 W and 1000 W. After the first layer of powder has

been completed, successive offset layering and remelting are continued until the porous part has been successfully completed. In this embodiment, the benefit of the part formed is that that decreased density increases porosity thus enabling  
5 the part to be used for, among other things, tissue ingrowth.

In a second embodiment, the first layer of metallic powder is deposited onto a solid base or core and fused thereto. Preferred metals used for the core include titanium, titanium alloys, stainless steel, cobalt chrome alloys, tantalum and  
10 niobium. Successive powder layers of the same or different materials are once again added in a layer-by-layer fashion until the part is completed. This embodiment has the desired effect of providing a structure in which the porosity may be increased as the structure is built, resulting in a graded  
15 profile in which the mechanical properties will also be reduced outwards from the core. This will allow the formed part to be used for, among other things, medical implants and prosthesis, but yet still include a surface for tissue ingrowth.

The method of producing a three-dimensional porous tissue  
20 ingrowth structure may include depositing a first layer of a powder made from a metal selected from the group consisting of titanium, titanium alloys, stainless steel, cobalt chrome alloys, tantalum and niobium, onto a substrate. Followed by scanning a laser beam at least once over the first layer of  
25 powder. The laser beam having a power (P) in Joule per seconds with a scanning speed (v) in millimeters per second with a range between 80 and 400 mms. and a beam overlap (b) in millimeters of between 50% and -1200%. Such that the number calculated by the formula  $P/(b \times v)$  lies between the range  
30  $0.3-8 \text{ J} / \text{mm}^2$ .

At least one additional layer of powder is deposited and then the laser scanning steps for each successive layer are

repeated until a desired web height is reached. In a second embodiment, during the step of repeating the laser scanning steps, at least one laser scan is carried out angled relative to another laser scan in order to develop an interconnecting or  
5 non-interconnecting porosity.

The thickness of the first layer and said successive layers of powder is between 5 $\mu$ m-2000  $\mu$ m. The laser can be applied either continuously or in a pulse manner, with the frequency of the pulse being in the range of approximately  
10 1 KHz to 50 KHz. Preferably, the method is carried out under an inert atmosphere, more preferably specifically an Argon inert atmosphere.

In order to achieve a greater mechanical strength between the base or core and the first layer of powder a third metal  
15 may be used to act as an intermediate. The third metal would act as a bond coat between the core and first layer of powder. Additionally the core may be integral with the resultant porous ingrowth structure and impart additional physical properties to the overall construct. The core may also be detachable from  
20 the resultant porous surface buildup.

It is the object of the present invention to provide a method of fabricating porous and partially porous metallic structures with a known porosity for use in particularly but not exclusively hard or soft tissue interlock structures or  
25 medical prosthesis.

These and other objects are accomplished by a process of fabricating an article in which laser-directed techniques are used to produce a porous three-dimensional structure with interconnected porosity and predetermined pore density, pore  
30 size and pore-size distribution. The article is fabricated, in the example of remelting, by using a laser and varying either the power of the laser, the layer thickness of the powder,



laser beam diameter, scanning speed of the laser or overlap of the beam. In fabricating a three-dimensional structure, the powder can be either applied to a solid base or not. The article is formed in layer-wise fashion until completion.

5 BRIEF DESCRIPTION OF THE DRAWINGS

Methods of forming the porous surface of the present invention can be performed in many ways and some embodiments will now be described by way of example and with reference to the accompanying drawings in which:

10 FIG. 1 is a diagrammatic illustration of the apparatus used to make test samples according to the processes of the present invention;

15 FIG. 2 is a photographic image showing an array of samples produced by the processes as performed by the apparatus of FIG. 1;

FIG. 3 is a table showing a series of parameters used for the samples of FIG. 2;

20 FIGS. 4 to 10 are scanning electron microscope images of the surface structure of various samples made by the method according to the invention;

FIG. 11 is a scanning electron microscope micrograph taken from a porous Ti sintered structure;

FIG. 12 is an optical image of a section through a sample showing the microstructure;

25 FIG. 13 is an image detailing surface structures;

FIGS. 14 and 15 are non-contact surface profilometry images detailing plan views of the samples; and

30 FIGS. 16 to 25 are scanning electron microscope micrographs produced prior to multi-layer builds shown in FIGS. 7 and 8.

FIG. 26 indicates the metallography and spectra of a typical bond coat structure.

FIG. 27 shows the effect of line spacing on pore size.

FIG. 28a-f are examples of typical waffle structures.

FIG. 29 is a trabecular bone-type structure constructed from a micro CT scan.

5 FIG. 30 shows typical freestanding structures.

FIG. 31 shows a freestanding structure built using the preferred scanning strategy.

#### DETAILED DESCRIPTION OF THE INVENTION

The present invention relates to a method of forming  
10 porous and partially porous metallic structures which are particularly but not exclusively applicable for use in hard or soft tissue interlock structures for medical implants and prosthesis. The method makes use of laser technology by employing a variety of scanning strategies. Typical metal and  
15 metal alloys employed include stainless steel, cobalt chromium alloys, titanium and its alloys, tantalum and niobium, all of which have been used in medical device applications. The present invention can be used for such medical device applications where bone and soft tissue interlock with a  
20 component is required, or where a controlled structure is required to more closely match the mechanical properties of the device with surrounding tissue. Additionally, the present invention may be employed to enhance the biocompatibility of a porous structure with animal tissue. With these advantages in  
25 mind, a structure may be created using specific dimensions required to accommodate a particular patient.

One particular intention of the present invention is to produce a three-dimensional structure using a direct laser remelt process, for example, for building structures with or  
30 without a solid base or core. When applied to an orthopedic prosthesis, the three-dimensional structure could be used to provide a porous outer layer to form a bone in-growth

structure. Alternatively, the porous structure, when applied to a core, could be used to form a prosthesis with a defined stiffness to both fulfill the requirement of a modulus match with surrounding tissue and provide interconnected porosity for tissue interlock. A further use could be to form an all-porous structure with grade pore size to interact with more than one type of tissue. Again, the process can be used to build on a solid base or core with an outer porous surface, the porosity of which is constant or which varies. The base or core materials to which the process is applied is either titanium and its alloys, stainless steel, cobalt chrome alloys, tantalum or niobium. The preferred surface coatings are titanium, cobalt chrome and tantalum but both stainless steel and niobium can also be used. Fully porous structures may be built from any of the materials tested, with the preferred material being titanium. One intention of the present invention is to produce a method which can be exploited on a commercial basis for the production of, for example, bone interlock surfaces on a device although it has many other uses.

According to the present invention, a method of forming a three-dimensional structure includes building the shape by laser melting powdered titanium and titanium alloys, stainless steel, cobalt chrome alloys, tantalum or niobium. The laser may be a continuous wave or pulsed laser beam.

The method can be performed so that the structure is porous and if desired, the pores can be interconnecting to provide an interconnected porosity.

If desired, the method can include using a base or core of cobalt chrome alloy, titanium or alloy, stainless steel, niobium and tantalum, on which to build a porous layer of any one of the aforementioned metals and alloys by laser melting

using a continuous or pulsed laser beam. Thus, a mixture of desired mixed materials may be employed.

Thus, the method can be applied to an existing article made from cobalt chrome, titanium or titanium alloys, stainless steel, tantalum or niobium, such as an orthopedic implant, to produce a porous outer layer from any of the aforementioned metals or alloys to provide a bone in-growth structure.

Preferably, prior to the deposition of any powder onto a substrate, a cleaning operation to ensure a contaminant-free surface may be employed. Typically, this process may include a solvent wash followed by a cleaning scan of the laser beam without the presence of particles.

In order to increase the mechanical bond between a substrate i.e., core or base, and a first layer of deposited powder a coating process may be employed. The coating process includes applying a third metal directly to the substrate, which has a higher bond strength to the substrate than does the first layer of powder. This process is particularly useful when the substrate and first powder layer are of different materials. The process of coating the substrate may be accomplished using known processes including laser deposition, plasma coating, cold gas dynamic spraying or similar techniques. One example of the coating process includes using either niobium or tantalum as an interface between a cobalt chrome alloy substrate and a first layer of titanium powder.

The present invention can also include a laser melting process, which precludes the requirement for subsequent heat treatment of the structure, thereby preserving the initial mechanical properties of the core or base metal.

The present invention may be applied to produce an all-porous structure using any of the aforementioned metal or metal alloys. Such structures can be used as finished product or

further processed to form a useful device for either bone or soft tissue in-growth. Additionally, the structure may be used to serve some other function such as that of a lattice to carry cells.

5       The pore density, pore size and pore size distribution can be controlled from one location on the structure to another. It is important to note that successive powder layers can differ in porosity by varying factors used for laser scanning powder layers. As for example, a first layer of powder is  
10 placed and subsequently scanned. Next a second layer of powder is placed and scanned. In order to control porosity the second scan may be angled relative to the first scan. Additionally, the angling of the scanning as compared to previous and post scans may be maneuvered and changed many times during the  
15 process of building a porous structure. If a structure was built without alternating the angling of any subsequent scans you would produce a structure having a plurality of walls rather than one with an interconnecting or non-interconnecting porosity.

20       In one such method, the laser melting process includes scanning the laser beam onto the powder in parallel scan lines with a beam overlap i.e., scan spacing, followed by similar additional scans or subsequent scans at 90°. The type of scan chosen may depend on the initial layer thickness as well as the  
25 web height required. Web height refers to the height of a single stage of the porous structure. The web height may be increased by deposited additional layers of powder of a structure and scanning the laser at the same angle of the previous scan.

30       Further, the additional scan lines may be at any angle to the first scan, to form a structure with the formation of a defined porosity, which may be regular or random. The scan

device may be programmed to proceed in a random generated manner to produce an irregular porous construct but with a defined level of porosity. Furthermore, the scan can be pre-programmed using digitized images of various structures, such as a trabecular bone, to produce a similar structure. Contrastingly, the scan may be pre-programmed using the inverse of digitized images, such as the inverse of a digitized trabecular bone to produce trabecular shaped voids. Many other scanning strategies are possible, such as a waffle scan, all of which can have interconnecting porosity if required.

The beam overlap or layer overlap may be achieved by rotation of the laser beam, the part being produced, or a combination of both.

A first method according to the present invention is intended to produce a porous structure for bone in-growth on the outer surface layer of a prosthesis.

To produce a porous surface structure, the nature of the material formed as a result of laser melting of powdered beads is principally dependent on the thermal profile involved (heating rate, soaking time, cooling rate); the condition of the raw material (size and size distribution of powder particles); atmospheric conditions (reducing, inert or oxidizing chamber gas); and accurate control of the deposited layer thickness.

There have been a number of studies to determine the optimum pore structure for maximization of bone in-growth on prostheses. The general findings suggest that optimum porosity is between approximately 20% and 40%, and aim to mid value with a mean volume percent of voids of about 70%. The preferred pore structure is irregular and interconnected, with a minimum pore size between about 80  $\mu\text{m}$  and 100  $\mu\text{m}$  and a maximum pore size between 80  $\mu\text{m}$  and 800  $\mu\text{m}$ . The structured thickness for

in-growth is 1.4-1.6mm, but can be larger or smaller depending on the application. As for example, it may be necessary to produce even smaller pore sizes for other types of tissue interaction or specific cellular interaction.

5       The first phase of development of the present invention involved an investigation, designed to characterize the material transformation process and to identify the optimum parameters for processing using three substrate materials CoCr and Ti stainless steel alloys, with five powder types Ti, CoCr,  
10   Ta and Nb, stainless steel.

The initial Direct Laser Remelting trials explored a comprehensive range of process parameters during the production of a number of coated base substrates. Results from this task were evaluated using laser scanning and white light  
15   interferometry in order to define the range of process parameters that produced the optimum pore structure.

Referring to FIG. 1, there is shown the apparatus used to carry out the method which comprises an Nd; YAG industrial laser 10 manufactured by Rofin Sinar Lasers, in Hamburg,  
20   Germany, integrated to an RSG1014 analogue galvo-scanning head 12 providing a maximum scan speed of 500 mm/s. The laser beam 14 is directed into an atmospherically controlled chamber 16, which consists of two computer-controlled platforms for powder delivery and part building. The powder is delivered from a  
25   variable capacity chamber 18 into the chamber 16 and is transported by a roller 20 to a build platform 22 above a variable capacity build chamber 24. In the embodiment shown in Fig. 1, the build and delivery system parameters are optimized for an even 100  $\mu$ m coating of powder to be deposited for every  
30   build layer. The metals chosen as surface materials are all difficult to process due to their affinity for oxygen. Cr and Ti are easily oxidized when processed by laser in oxygen-

containing atmosphere, their oxide products have high melting points and poor flowability. For this reason, and to prevent the formation of other undesirable phases, the methods were carried out under an Argon inert atmosphere in chamber 16.  
5 Pressure remained at or below atmospheric pressure during the entire application.

The build chamber 24 illustrated in Fig. 1 and method of layering described above is suitable for test specimens and constructs such as three-dimensional freestanding structures.  
10 However, in order to build on to an existing device, such as acetabular metal shells, hip and knee femoral components, knee tibial components and other such devices, considerable changes to the powder laying technique would need to be applied.

Preliminary experiments were performed on CoCr alloy to  
15 determine the efficacy of in-situ laser cleaning of the substrate. Typical processing conditions were: Laser power of 82 W, pulse frequency of 30 KHz, and a laser scan speed of 160 mm/sec.

Preliminary experiments were performed on CoCr to assess  
20 the environment conditions within the chamber. In these examples, Co212-e Cobalt Chrome alloy was used. The CoCr was configured into square structures, called coupons. Arrays of CoCr coupons were built onto a stainless steel substrate. The Co212-e Cobalt Chrome alloy had a particle size distribution of  
25 90<22 um, i.e., 90% of the particles are less than 22 um, the composition of which is shown in the table below.

Table 1

Composition of Co212-e CoCr alloy

Element	Cr	Mo	Si	Fe	Mn	Ni	N	C	Co
Wt%	27.1	5.9	0.84	0.55	0.21	0.20	0.16	0.050	Balance



An array of nine sample coupons were produced as shown in FIG. 2, with the process of Table 2, using a maximum laser power of 78 watts (W) and laser scanning speed for each coupon varying between 100-260 mms<sup>-1</sup>. Of course a higher laser power may be employed; however, a higher laser power would also necessitate increasing the speed of the laser scan speed in order to produce the desired melting of the powder layer. A simple linear x-direction scan was used on each of the coupons. This allowed the processing parameter, beam overlap, to be used to control the space between successive scan lines. That is, with a 100 μm laser spot size, an overlap of -200% produces a 100 μm gap between scans. Although the acceptable range for the beam overlap is given at +50% to -1200% it should be duly noted that the negative number only refers to the fact there is a gap as opposed to a beam overlap between successive scans. For instance a beam overlap of zero refers to the fact that successive scans on the same layer of powder border each other. If the beam overlap was 5% then 5% of the first scan is overlapped by the second scan. When computing the Andrew number the absolute value of the beam overlap is used. The complete set of process parameters used is shown in Table 2 below.

Table 2 Process parameters

Power Watts (W)	Layer Thickness (μm)	Beam Diameter (μm)	Scanning Speed (mms <sup>-1</sup> )	Atmosphere	No. of Layers	Overlap (% of line width)
78	100	100	100-260	No	16	25, 50, -500

The incremental changes in scanning speed and the size of the speed range were modified as the experiments progressed. To begin with, a large range of speeds was used to provide an

initial indication of the material's performance and the propensity to melt. As the experiments progressed, the range was reduced to more closely define the process window. Speed and beam overlap variations were used to modify the specific energy density being applied to the powder bed and change the characteristics of the final structure. The complete series of parameters are given in FIG. 3, the parameters sets used for the definitive samples are shaded in gray.

CoCr was the first of four powders to be examined and, therefore, a wide range of process parameters was used. In each case, laser power and the pulse repetition rate were kept constant, i.e., continuous laser pulse, to allow the two remaining parameters to be compared. Layer thickness was maintained at 100  $\mu\text{m}$  throughout all the experiments described here. Layer thickness can, however, vary between 5  $\mu\text{m}$  to 2000  $\mu\text{m}$ .

On completion of the initial series of experiments using CoCr powder on 2.5 mm thick stainless steel substrates, basic optical analysis was conducted of the surface of the coupons to ascertain the size of the pores and degree of porosity being obtained. Once a desired pore size was obtained and the coupons had suitable cohesion, the two experiments closest to the optimum desired pore size were repeated using first CoCr and then Ti substrates. The same structure could be obtained by other parameters.

Following the conclusion of the CoCr experiments, the remaining three powders; Niobium, Tantalum and Titanium were investigated in turn. The procedure followed a simple course although fewer parameter sets were explored as the higher melting points of these materials forced the reduction in speeds compared to CoCr powder.

For Niobium, the particle size description was 80% < 75  $\mu\text{m}$  at a purity of 99.85%. Due to its higher melting temperature compared to that of CoCr (Nb being at about 2468°C, and CoCr being at about 1383°C), the laser parameters used included a  
5 reduced scanning speed range and increased beam overlap providing increased specific energy density at the powder bed. In addition, the pulse repetition rate was varied from 20 kHz to 50 kHz.

On completion of a small number (four in total) of  
10 preliminary experiments of Nb on stainless steel substrate, the experiment with the most ideal parameters was repeated on both the CoCr and Ti substrates.

The Tantalum used in this study had a particular size distribution of 80% < 75  $\mu\text{m}$  with a purity of 99.85%. Ta has a  
15 melting point of about 2996°C and was processed using the same laser parameters as Nb. Now confident of the atmospheric inertness, the Ta powder was melted directly onto the CoCr and Ti substrates.

Bio-medical alloys of Titanium were not readily available  
20 in powder form and so pure Ti was chosen. The particle size distribution for the Ti powder was 80% < 45  $\mu\text{m}$  with a purity of 99.58%. The same parameters used for Nb and Ta were also used for the Ti powder. Ti has a lower melting point than Ta or Nb, Ti being at about 1660°C, but has a higher thermal conductivity  
25 than Ta or Nb. This implies that although the powder should require less energy before melting, the improved heat transfer means a larger portion of the energy is conducted away from the melt pool.

Following the completion of samples with all four powders  
30 on the required substrates, surface analysis was conducted using optical analysis and a scanning electron microscope to obtain images of the resultant pores. Porosity was calculated

using a simple image processing technique involving the setting of contrast thresholds and pixel counting. While this technique is not the most accurate method, it allows the rapid analysis of small samples produced. Techniques such as Xylene  
5 impregnation would yield more accurate results but they are time consuming and require larger samples than those produced here.

Following an extended series of experimentation, two sets of laser processing parameters were selected for the laser  
10 melting of CoCr powder. From analysis of the stainless steel substrates, it was seen that a large portion of the results fell within the pore size required for these materials, stated as being in the range of 80  $\mu\text{m}$  to 400  $\mu\text{m}$ .

Optical analysis of the surface structure of each of the  
15 coupons produced with CoCr on CoCr and Ti alloy substrates were initially viewed but due to problems with the depth of field associated with an optical microscope, little information could be extracted. In addition to the coupons that were produced to satisfy the project requirements, two experiments were  
20 conducted using a relatively large negative beam overlap of -250 and -500%. Optical images of the coupon's surface and in section are shown in FIG. 4. These were not the definitive parameters chosen for the final arrays on CoCr and Ti alloy substrates as the pore size exceeds the required 80  $\mu\text{m}$  to  
25 400  $\mu\text{m}$ . They are shown here to display what the Direct Laser Remelting process can produce when an excessive beam overlap is used.

To provide a clearer indication of the pore size and pore density, the optical analysis was repeated using images  
30 obtained from the scanning electron microscope. FIG. 5 is an image of two coupons produced from a CoCr array on Ti alloy substrates. This array was chosen because it best satisfied

the requirements of this exercise. The parameters were: laser power of 82 W continuous wave (cw); 25% beam overlap; scanning speed varied from 100 mms<sup>-1</sup> to 260 mms<sup>-1</sup> in 20mm<sup>-1</sup> increments; the images of the coupons shown here, taken from this array, were produced with scanning speeds of 180 mms<sup>-1</sup> to 200 mms<sup>-1</sup>. The surface is comprised of fused pathways that develop a network of interconnected pores. This structure continues throughout the layer until the interface is reached. The interface is characterized by a patchwork of fusion bonds. These bond sites are responsible for securing the interconnected surface structure to the baseplate. The macroscopic structure is covered with unmelted powder particles that appear to be loosely attached. In addition, there are larger resolidified globules that may have limited bonding to the surface.

FIGS. 6 and 7 are the scanning electron microscope images produced from the Nb and Ta coupons on Ti alloy substrates. Specifically, FIGS. 6(a) to 6(e) are scanning electron microscope images of the surface structure of Nb on Ti alloy substrates, produced with a laser power of 82 W cw, -40% beam overlap. The scanning speeds used were: 160 mms<sup>-1</sup> for FIG. 6(a), 190 mms<sup>-1</sup> for FIG. 6(b), 200 mms<sup>-1</sup> for FIG. 6(c), 210 mms<sup>-1</sup> for FIG. 6(d) and 240 mms<sup>-1</sup> for FIG. 6(e), respectively.

FIGS. 7(a) to 7(c) are scanning electron microscope images of the surface structure of Ta on Ti alloy substrates produced using the same parameters used in the Nb or Ti alloy substrates except: FIG. 7(a) was produced with a scanning speed of 160 mms<sup>-1</sup>; FIG. 7(b)'s speed was 200 mms<sup>-1</sup> and FIG. 7(c)'s speed was 240 mms<sup>-1</sup>, respectively. An increased beam overlap was used here as Nb and Ta have high melting points, which require a greater energy density. The surfaces once again exhibit significant levels of unmelted powder particles and loosely attached

resolidified beads that vary in size from a few microns to several hundred microns. All samples were loosely brushed after completion and cleaned in an ultrasonic aqueous bath. It is possible that further cleaning measures may reduce the  
5 fraction of loose particles.

FIGS. 8(a) to 8(e) are scanning electron microscope images taken from the Ti coupons on the CoCr alloy substrates. The laser processing parameters used were the same as those for the Nb and Ta powders, with once again only the speed varying. The  
10 scanning speed was varied from 160  $\text{mms}^{-1}$  (FIG. 8(a)), 170  $\text{mms}^{-1}$  (FIG. 8(b)), 200  $\text{mms}^{-1}$  (FIG. 8(c)); 230  $\text{mms}^{-1}$  (FIG. 8(d) to 240  $\text{mms}^{-1}$  (FIG. 8(e)). The Ti coupon on CoCr samples, (FIGS. 8(a) to 8(c)) indicate very high density levels compared to the other examples. The line-scans can be clearly seen, with good  
15 fusion between individual tracks, almost creating a complete surface layer. The surface begins to break up as the scanning speed is increased.

FIGS. 9(a) to 9(e) are scanning electron microscope images of surface structures of Ti on Ti alloy substrates produced  
20 with the same parameters used in FIGS. 8(a) to 8(e), respectively. It is unclear why Ti should wet so well on CoCr substrates. In comparison, Ti on Ti exhibits similar characteristic patterns as with Nb, Ta, and CoCr, specifically, an intricate network of interconnected pores.

25 Following the completion of the multi-layer coupons, a series of 20mm x 20mm structures were produced from Ti that utilized an X and Y-direction "waffle" scanning format using the optimum Ti operating parameters with the two scans being orthogonal to one another. The intention behind these  
30 experiments was to demonstrate the ability of the Direct Laser Remelting process to produce parts with a controlled porosity, e.g. internal channels of dimensions equal to the required pore

size, e.g. 80  $\mu\text{m}$  to 400  $\mu\text{m}$ . To do this, a relatively large beam overlap of between -400% and -600% was used. Scanning electron microscope images of the surfaces of these structures are shown in FIGS. 10(a) to 10(f). The scanning speed, 5 160  $\text{mms}^{-1}$  and the laser power 72 W cw were kept constant while the beam overlaps; -400% in FIGS. 10(a) and 10(b); -500% in FIGS. 10(c) and 10(d) and -600% in FIGS. 10(e) and 10(f), were varied accordingly. Scanning electron microscope micrographs, taken from a porous Ti sintered structure provided by 10 Stryker-Howmedica are shown for reference in FIG. 11.

To illustrate more clearly the interaction between the substrate/structure metallurgical interaction, the Ti on Ti substrate was sectioned, hot mounted and polished using a process of 1200 and 2500 grade SiC, 6  $\mu\text{m}$  diamond paste and 15 70/30 mixture of OPS and 30%  $\text{H}_2\text{O}_2$ . The polished sample was then etched with 100ml  $\text{H}_2\text{O}$ , 5ml  $\text{NH}_4\text{FHF}$  and 2 $\text{cm}^3$   $\text{HCl}$  for 30 seconds to bring out the microstructure. Optical images of this sample in section are shown in FIG. 12.

FIG. 13 is an image taken from a non-contact surface 20 profilometry to show the surface structures obtained when using Ti, CoCr, Ta and Nb on Ti substrates. Values for  $R_a$ ,  $R_q$  and  $R_b$  roughness are also shown.

From the optical and scanning electron microscope analysis conducted, it is apparent that the Direct Laser Remelting 25 process is capable of satisfying the requirements for pore characteristics, concerning maximum and minimum pore size, interconnectivity and pore density. From the initial visual analysis of the CoCr coupons, it was apparent from these and other examples, that subtle variations in pore structure and 30 coverage could be controlled by scanning velocity and line spacing.

The key laser parameters varied for forming the three-dimensional metallic porous structures are: (a) Laser scanning speed (v.) in (mms<sup>-1</sup>), which controls the rate at which the laser traverses the powder bed; (b) Laser power, P(W), which in conjunction with the laser spot size controls the intensity of the laser beam. The spot size was kept constant throughout the experiment; (c) Frequency, (Hz) or pulse repetition rate. This variable controls the number of laser pulses per second. A lower frequency delivers a higher peak power and vice versa.

The line width can be related to the laser scanning speed and the laser power to provide a measure of specific density, known as the "Andrew Number", where:

$$An = \frac{P}{b \times v} \quad (J/mm^{-2})$$

15

Where P denotes the power of the laser, v is the laser scanning speed and b denotes beam width of the laser. The Andrew number is the basis for the calculation of the present invention. The Andrew number may also be calculated by substituting the line separation (d) for beam width (b). The two methods of calculating the Andrew number will result in different values being obtained. When using line separation (d) as a factor only one track of fused powder is considered, whereas when using the beam width (b) as a factor, two tracks of fused powder are considered as well as the relative influence of one track to the next. For this reason we have chosen to concern ourselves with the Andrew number using scan spacing as a calculating factor. It can thus be appreciated, that the closer these tracks are together the greater the influence they have on one another.

30

Additionally, the laser power may be varied between 5 W and 1000 W. Utilizing lower power may be necessary for small



and intricate parts but would be economically inefficient for such coatings and structures described herein. It should be noted that the upper limit of laser power is restricted because of the availability of current laser technology. However, if a  
5 laser was produced having a power in excess of 1000 W, the scanning speed of the laser could be increased in order that an acceptable Andrew number is achieved. A spot size having a range between 5 um(fix) to 500 um(fix) is also possible. For the spot size to increase while still maintaining an acceptable  
10 Andrew number, either the laser power must be increased or the scanning speed decreased.

The above formula gives an indication of how the physical parameters can vary the quantity of energy absorbed by the powder bed. That is, if the melted powder has limited  
15 cohesion, e.g. insufficient melting, the parameters can be varied to concentrate the energy supply to the powder. High Andrew numbers result in reduced pore coverage and an increase in pore size due to the effects of increased melt volume and flow. Low Andrew numbers result in low melt volume, high pore  
20 density and small pores. Current satisfactory Andrew numbers are approximately  $.3 \text{ J/mm}^{-2}$  to  $8 \text{ J/mm}^{-2}$  and are applicable to many alternative laser sources. It is possible to use a higher powered laser with increased scanning speed and obtain an Andrew number within the working range stated above.

25 Line spacing or beam overlap can also be varied to allow for a gap between successive scan lines. It is, therefore, possible to heat selected areas. This gap would allow for a smaller or larger pore size to result. The best illustration of this is shown in FIGS. 4(a) to 4(c) where a -500% beam  
30 overlap has been applied. FIGS. 4(a) to 4(c) are scanning electron microscope images of the surface structure of CoCr on stainless steel produced with a laser power of 82 W cw. FIG.

4(a) was produced with a laser scanning speed of  $105 \text{ mms}^{-1}$  and FIG. 4(b) was produced with a laser scanning speed of  $135 \text{ mms}^{-1}$ . FIG. 4(c) is an image of the same structure in FIG. 4(b), in section. There is a significant self-ordering within the overall structure. Larger columnar structures are selectively built leaving large regions of unmelted powder. It is worth noting that these pillars are around  $300 \mu\text{m}$  wide, over  $1.6 \text{ mm}$  tall and fuse well with the substrate, as seen in FIG. 4(c). Further analysis shows that the use of a hatched scanning format allows porosity to be more sufficiently controlled to allow the pore size to be directly controlled by the beam overlap.

The use of an optical inspection method to determine this approximate porosity is appropriate given the sample size. This method, although not accurate due to the filter selection process, can, if used carefully, provide an indication of porosity. An average porosity level of around 25% was predicted. This porosity level falls within the range of the desired porosity for bone in-growth structures. The mechanical characteristics of the porous structures are determined by the extent of porosity and the interconnecting webs. A balance of these variables is necessary to achieve the mechanical properties required by the intended application.

Increased fusion may, if required, be obtained by heating the substrate, powder or both prior to scanning. Such heating sources are commonly included in standard selective laser sintering/melting machines to permit this operation.

Following trials on the titanium build on the cobalt chromium substrate, it was determined that the interface strength was insufficient to serve the intended application. Trials were made by providing a bond coat of either tantalum or niobium on the cobalt chromium substrate prior to the

deposition of the titanium layers to for the porous build. The typical protocol involved:

- (i) Initial cleaning scan with a scan speed between 60 to 300 mm/sec, laser power 82 watts, frequency of 30KHz, and a 50% beam overlap.
- (ii) Niobium or tantalum deposition with three layers of 50µm using a laser power of 82 watts, frequency 30 to 40KHz, with a laser speed of between 160 to 300 mm/sec. The beam overlap was low at 50% to give good coverage.
- (iii) A build of porous titanium was constructed using a laser power of 82 watts, frequency between 0 (cw) and 40KHz, scanning speed of between 160 and 240 mm/sec, and beam overlap of -700%.

The strengths of the constructs are indicated in Table 3 with a comparison of the values obtained without the base coat.

5

Table 3

Specimen	Maximum load (kN)	Tensile Strength (MPa)	Failure mode
Ti on CoCr	2.5	5	Interface
Ti on CoCr	3.1	6.2	Interface
1 (Nb on Co-Cr)	13.0	26.18	65% adhesive, 35% bond interface
4 (Ti on Nb onCo-Cr)	7.76	15.62	Mostly bond coat interface
5 (Ti on Nb onCo-Cr)	9.24	18.53	20% adhesive, 40% bond coat, 40% porous Ti
6 (Ti on Ta onCo-Cr)	11.58	23.33	Mostly adhesive with discrete webbing weakness
8 (Ta onCo-Cr)	13.93	27.92	60% adhesive, 40% bond interface
9 (Ti on Ta onCo-Cr)	6.76	13.62	100% bond interface

FIG. 26 shows the metallography of the structures with energy dispersive spectroscopy (EDS) revealing the relative metal positions within the build.

A typical waffle build of titanium on a titanium substrate was constructed as a way of regulating the porous structure. Scanning sequences of  $0^\circ\ 0^\circ\ 0^\circ$ ,  $90^\circ\ 90^\circ\ 90^\circ$ ,  $45^\circ\ 45^\circ\ 45^\circ$ ,  $135^\circ, 135^\circ, 135^\circ$ , of layer thickness 0,1mm were developed to form a waffle. Three layers of each were necessary to obtain sufficient web thickness in the "z" direction to give a structure of adequate strength. Typical parameters employed were: Laser power was 82 watts, operating frequency between 0 (cw) and 40KHz, scan speed of between 160 and 240 mm/sec with a beam overlap of -700%. FIG. 27 gives an indication of the effect of line spacing and pore size. FIG. 28 (a) shows typical examples of the waffle structure. The magnification level changes from 10, 20, 30, 70 and 150 times normal viewing as one moves respectively from Fig. (b) to (f). Figure 28 (a) more specifically shows Ti powder on a Ti substrate with a controlled porosity by varying line spacing, i.e., beam overlap.

Trabecular structures of titanium on a titanium substrate were constructed as a way of randomising the porous structures. An STL (sterolithography) file representing trabecular structure was produced from a micro CT scan of trabecular bone. This file was sliced and the slice data sent digitally to the scanning control. This allowed the layer-by-layer building of a metallic facsimile to be realised. FIG. 29 shows a cross-sectional view of such a construct.

A method for making lattice-type constructs was referred to in the relevant art. A typical example of this type of structure is shown in FIG 30. The scanning strategy employed to form such a construct was mentioned and such a strategy

could be produced within the range of Andrew numbers outlined. Table 4 shows an indication of scanning strategies and their relationships to the Andrew number.

Table 4

5 Ti on Ta on CoCr Experimental Procedure.

**Initial Tantalum Coating**

Layer	Scan Strategy	Parameter Set	Layer Thickness	Relative Build Platform Position	Additional
Zero Distance Between Roller & Build Platform				0	
0	1 <sup>st</sup> layer thickness set using feeler gauges but powder not laid in preparation for cleaning scan with no powder.		50 $\mu$ m	-50 $\mu$ m	
1	50% Beam Overlap	P = 82W Qs = 30kHz v = 60mm/s A <sub>n</sub> = 27.333J/mm <sup>2</sup>			Initial Cleaning Scan (no powder)
	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>			Powder laid as usual
	50% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 5.467J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.

2	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>	50μm	-100μm	Powder laid as usual
	50% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 5.467J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.
3	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>	50μm	-150μm	Powder laid as usual
	50% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 5.467J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.

### Final Titanium Coating

Layer	Scan Strategy	Parameter Set	Layer Thickness	Relative Build Platform Position	Additional
0	1 <sup>st</sup> layer thickness set using feeler gauges but powder not laid in preparation for cleaning scan with no powder.			-150μm	
1	50% Beam Overlap	P = 82W Qs = 30kHz v = 60mm/s A <sub>n</sub> = 27.3J/mm <sup>2</sup>	50μm	-200μm	Cleaning Scan (No powder).
	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>			Powder spread but build platform not lowered.

	50% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 5.467/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.
2	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>	100μm	-300μm	Powder laid as usual
	25% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 3.644J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.
3	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>	100μm	-400μm	Powder laid as usual
	0% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 2.733J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.
4	Waffle 0 and 90° 700μm linespacing (-600% Beam overlap)	P = 82W Qs = 0Hz (cw) v = 240mm/s A <sub>n</sub> = 0.488J/mm <sup>2</sup>	75μm	-475μm	Powder laid as usual
5	Waffle 0 and 90° 700μm linespacing (-600% Beam overlap)	P = 82W Qs = 0Hz (cw) v = 240mm/s A <sub>n</sub> = 0.488J/mm <sup>2</sup>	75μm	-550μm	Powder laid as usual
6	Waffle 0 and 90° 700μm linespacing (-600% Beam overlap)	P = 82W Qs = 0Hz (cw) v = 240mm/s A <sub>n</sub> = 0.488J/mm <sup>2</sup>	75μm	-625μm	Powder laid as usual
7	Waffle 45 and 135° 700μm linespacing	P = 82W Qs = 0Hz (cw) v = 240mm/s	75μm	-700μm	Powder laid as usual

	(-600% Beam overlap)	$A_n = 0.488\text{J/mm}^2$			
8	Waffle 45 and 135° 700µm linespacing (-600% Beam overlap)	P = 82W Qs = 0Hz (cw) v = 240mm/s $A_n = 0.488\text{J/mm}^2$	75µm	-775µm	Powder laid as usual
9	Waffle 45 and 135° 700µm linespacing (-600% Beam overlap)	P = 82W Qs = 0Hz (cw) v = 240mm/s $A_n = 0.488\text{J/mm}^2$	75µm	-850µm	Powder laid as usual

**Ti on Ti Experimental Procedure.**

**Initial Titanium Coating**

5

Layer	Scan Strategy	Parameter Set	Layer Thickness	Relative Build Platform Position	Additional
	Zero Distance Between Roller & Build Platform			0	
0	1 <sup>st</sup> layer thickness set using feeler gauges but powder not laid in preparation for cleaning scan with no powder.		50µm	-50µm	
1	50% Beam Overlap	P = 82W Qs = 30kHz v = 60mm/s $A_n = 27.333\text{J/mm}^2$			Initial Cleaning Scan (no powder)
	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s $A_n = 5.125\text{J/mm}^2$			Powder laid as usual



	50% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 5.467J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.
2	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>	50μm	-100μm	Powder laid as usual
	50% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 5.467J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.
3	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>	50μm	-150μm	Powder laid as usual
	50% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 5.467J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.

### Final Titanium Coating

Layer	Scan Strategy	Parameter Set	Layer Thickness	Relative Build Platform Position	Additional
1	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>	100μm	-250μm	Powder laid as usual
	50% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 5.467J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan

2	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>	100μm	-350μm	Powder laid as usual
	25% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 3.644J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.
3	Circular profile. 5 concentric circles, 0.1mm offset to negate effects of 'First Pulse'	P = 82W Qs = 40kHz V = 160mm/s A <sub>n</sub> = 5.125J/mm <sup>2</sup>	100μm	-450μm	Powder laid as usual
	0% Beam Overlap	P = 82W Qs = 30kHz v = 300mm/s A <sub>n</sub> = 2.733J/mm <sup>2</sup>			Scanned on same powder layer as previous profile scan.
4	Waffle 0 and 90° 700μm linespacing (-600% Beam overlap)	P = 82W Qs = 0Hz (cw) v = 240mm/s A <sub>n</sub> = 0.488J/mm <sup>2</sup>	75μm	-525μm	Powder laid as usual
5	Waffle 0 and 90° 700μm linespacing (-600% Beam overlap)	P = 82W Qs = 0Hz (cw) v = 240mm/s A <sub>n</sub> = 0.488J/mm <sup>2</sup>	75μm	-600μm	Powder laid as usual
6	Waffle 0 and 90° 700μm linespacing (600% Beam overlap)	P = 82W Qs = 0Hz (cw) v = 240mm/s A <sub>n</sub> = 0.488J/mm <sup>2</sup>	75μm	-675μm	Powder laid as usual
7	Waffle 45 and 135° 700μm linespacing (-600% Beam overlap)	P = 82W Qs = 0Hz (cw) v = 240mm/s A <sub>n</sub> = 0.488J/mm <sup>2</sup>	75μm	-750μm	Powder laid as usual
8	Waffle 45 and 135° 700μm linespacing	P = 82W Qs = 0Hz (cw) v = 240mm/s A <sub>n</sub> =	75μm	-825μm	Powder laid as usual

	linespacing (-600% Beam overlap)	0.488J/mm <sup>2</sup>			
9	Waffle 45 and 135° 700µm linespacing (-600% Beam overlap)	P = 82W Qs = 0Hz (cw) v = 240mm/s A <sub>n</sub> = 0.488J/mm <sup>2</sup>	75µm	-900µm	Powder laid as usual

The second and preferred approach uses a continuous scanning strategy whereby the pores are developed by the planar deposition of laser melted powder tracks superimposed over each other. This superimposition combined with the melt flow produces random and pseudorandom porous structures. The properties of the final structure, randomness, interconnectivity, mechanical strength and thermal response are controlled by the process parameters employed. One set of scanning parameters used was: Scanning sequences of 0° 0°0°, 90° 90° 90°, 45° 45° 45°, 135°, 135°, 135°, of layer thickness 0.1mm were developed to form a waffle. Three layers of each were necessary to obtain sufficient web thickness in the "z" direction. The array of sequences was repeated many times to give a construct of the desired height. Laser power was 82 watts, operating frequency between 0 (cw) and 40KHz, scan speed of between 160 and 240 mm/sec with a beam overlap of -700%. FIG. 32 shows such a construct.

The use of an optical inspection method to determine this approximate porosity is appropriate given the sample size. This method, although not accurate due to the filter selection process, can, if used carefully, provide an indication of porosity. An average porosity level of around 25% was predicted. This porosity level falls within the range of the desired porosity for bone in-growth structures.

In consideration of the potential application, it is important to minimize loose surface contamination and

demonstrate the ability to fully clean the surface. Laser cleaning or acid etching technique may be effective. Additionally, a rigorous cleaning protocol to remove all loose powder may entail blowing the porous structure with clean dry compressed gas, followed by a period of ultrasonic agitation in a treatment fluid. Once dried, a laser scan may be used to seal any remaining loose particles.

On examination, all candidate materials and substrates were selectively fused to produce a complex interconnected pore structure. There were small differences in certain process parameters such as speed and beam overlap percentage. From FIG. 12 it can also be seen how the Ti build has successfully fused with the Ti alloy substrate using a laser power of 82 W cw, beam overlap of -40% and a laser scanning speed of 180 mms<sup>-1</sup>. With the ability to produce structures with a controlled porosity, this demonstrates how the Direct Laser Remelting process can be used as a surface modification technology. Certain metal combinations interacted unfavourably and resulted in formation of intermetallics, which are inherently brittle structures. To overcome this problem it may be necessary to use a bond coat with the substrate. It is then possible to build directly on to the substrate with a porous build. A typical example of the use of a bond coat is provided by the combination of titanium on to a cobalt chromium substrate. Tantalum also was an effective bond coat in this example. Note that the bond coat may be applied by laser technology, but other methods are also possible such as gas plasma deposition.

The non-contact surface profilometry (OSP) images shown in FIGS. 13(a) to 13(d) show the surface profile. In addition, the Surface Data shown in FIGS. 14(a) and 14(b) and 15(a) and 15(b) show a coded profile of the plan views of the samples.

FIG. 14(a) shows Ti on Ti (OSP Surface Data) where  $v = 200$   $\text{mms}^{-1}$ , FIG. 14(b) shows CoCr on Ti (OSP Surface Data) where  $v = 200$   $\text{mms}^{-1}$ , and FIG. 15(a) shows Nb on Ti (OSP Surface Data) where  $v = 200$   $\text{mms}^{-1}$  and FIG. 15(b) shows Ta on Ti (OSP Surface Data) where  $v = 200$   $\text{mms}^{-1}$ .

FIGS. 16 to 25 are scanning electron microscope (SEM) micrographs of a series of single layer Ti on CoCr and Ti on Ti images that were produced prior to the multi-layer builds shown in FIGS. 8 and 9 respectively and as follows.

10 FIG. 16(a) shows Ti on CoCr (Single Layer; SEM Micrograph)  
 $v = 160$   $\text{mms}^{-1}$ ;

FIG. 16(b) shows Ti on CoCr (Single Layer; SEM Micrograph)  
 $v = 160$   $\text{mms}^{-1}$ ;

15 FIG. 17(a) shows Ti on CoCr (Single Layer; SEM Micrograph)  
 $v = 170$   $\text{mms}^{-1}$ ;

FIG. 17(b) shows Ti on CoCr (Single Layer; SEM Micrograph)  
 $v = 180$   $\text{mms}^{-1}$ ;

FIG. 18(a) shows Ti on CoCr (Single Layer; SEM Micrograph)  
 $v = 190$   $\text{mms}^{-1}$ ;

20 FIG. 18(b) shows Ti on CoCr (Single Layer; SEM Micrograph)  
 $v = 200$   $\text{mms}^{-1}$ ;

FIG. 19(a) shows Ti on CoCr (Single Layer; SEM Micrograph)  
 $v = 210$   $\text{mms}^{-1}$ ;

25 FIG. 19(b) shows Ti on CoCr (Single Layer; SEM Micrograph)  
 $v = 220$   $\text{mms}^{-1}$ ;

FIG. 20(a) shows Ti on CoCr (Single Layer; SEM Micrograph)  
 $v = 230$   $\text{mms}^{-1}$ ;

FIG. 20(b) shows Ti on CoCr (Single Layer; SEM Micrograph)  
 $v = 240$   $\text{mms}^{-1}$ ;

30 FIG. 21(a) shows Ti on Ti (Single Layer; SEM Micrograph)  
 $v = 160$   $\text{mms}^{-1}$ ;

FIG. 21(b) shows Ti on Ti (Single Layer; SEM Micrograph) v  
= 170 mms<sup>-1</sup>;

FIG. 22(a) shows Ti on Ti (Single Layer; SEM Micrograph) v  
= 190 mms<sup>-1</sup>;

5 FIG. 22(b) shows Ti on Ti (Single Layer; SEM Micrograph) v  
= 200 mms<sup>-1</sup>;

FIG. 23(a) shows Ti on Ti (Single Layer; SEM Micrograph) v  
= 220 mms<sup>-1</sup>;

10 FIG. 23(b) shows Ti on Ti (Single Layer; SEM Micrograph) v  
= 230 mms<sup>-1</sup>;

FIG. 24(a) shows Ti on Ti (Single Layer; SEM Micrograph) v  
= 240 mms<sup>-1</sup>;

FIG. 24(b) shows Ti on Ti (Single Layer; SEM Micrograph) v  
= 240 mms<sup>-1</sup>;

15 The method according to the present invention can produce  
surface structures on all powder/baseplate combinations with  
careful selection of process parameters.

As described above, the process is carried out on flat  
baseplates that provide for easy powder delivery in successive  
20 layers of around 100  $\mu\text{m}$  thickness. Control of powder layer  
thickness is very important if consistent surface properties  
are required. The application of this technology can also be  
applied to curved surfaces such as those found in modern  
prosthetic devices; with refinements being made to the powder  
25 layer technique.

The structures have all received ultrasonic and aqueous  
cleaning. On close examination, the resultant porous surfaces  
produced by the Direct Laser Remelting process exhibit small  
particulates that are scattered throughout the structure. It  
30 is unclear at this stage whether these particulates are bonded  
to the surface or loosely attached but there are means to  
remove the particulates if required.

The Direct Laser Remelting process has the ability to produce porous structures that are suitable for bone in-growth applications. The powdered surfaces have undergone considerable thermal cycling culminating in rapid cooling rates that have produced very fine dendritic structures (e.g. FIGS. 13(a) to 13(d)).

The Direct Laser Remelting process can produce effective bone in-growth surfaces and the manufacturing costs are reasonable.

In the preceding examples, the object has been to provide a porous structure on a base but the present invention can also be used to provide a non-porous structure on such a base to form a three-dimensional structure. The same techniques can be utilized for the materials concerned but the laser processing parameters can be appropriately selected so that a substantially solid non-porous structure is achieved.

Again, a technique can be used to deposit the powder onto a suitable carrier, for example a mold, and to carry out the process without the use of a base so that a three-dimensional structure is achieved which can be either porous, as described above, or non-porous if required.

It will be appreciated that this method can, therefore, be used to produce articles from the metals referred to which can be created to a desired shape and which may or may not require subsequent machining. Yet again, such an article can be produced so that it has a graded porosity of, e.g., non-porous through various degrees of porosity to the outer surface layer. Such articles could be surgical prostheses, parts or any other article to which this method of production would be advantageous.

CLAIMS:

1. A method of producing a three-dimensional porous tissue ingrowth structure comprising:

5 depositing a first layer of a powder made from a metal selected from the group consisting of titanium, titanium alloys, stainless steel, cobalt chrome alloys, tantalum and niobium, onto a substrate;

10 scanning a laser beam at least once over said first layer of powder, said laser beam having a power (P) in Jovle per sec. with a scanning speed (v) in millimeters per sec., and a beam overlap (b) in millimeters such that the number calculated by the formula  $P/(b \times v)$  lies between the range 0.3-8 J / mm<sup>2</sup>, said beam overlap being approximately between +50% to -1200% to give the required pore size;

15 depositing at least one layer of said powder onto said first layer; and

repeating said laser scanning steps for each successive layer until a desired web height is reached.

20 2. The method according to claim 1, wherein during said step of repeating said laser scanning steps, at least one laser scan is carried out angled relative to another laser scan in order to develop an interconnecting or non-interconnecting porosity.

25 3. The method according to claim 1, wherein the thickness of each of said first layer and said successive layers of powder is between 5 $\mu$ m -2000  $\mu$ m.

30 4. The method according to claim 1, wherein said substrate is a base or core made of a metal selected from the group consisting of titanium, titanium alloys, stainless steel, cobalt chrome alloys, tantalum and niobium, wherein said first layer is fused to said base or core.



5. The method according to claim 1, wherein said laser power may be varied within a range between 5 to 1000 watts.

6. The method according to claim 1, wherein said laser power is applied in continuously.

7. The method according to claim 1, wherein said laser power is applied in a pulsed manner.

8. The method according to claim 7, wherein said pulse has a repetition rate between approximately 1 KHz to 50 KHz.

9. The method according to claim 1, wherein the method is carried out under an inert atmosphere.

10. The method according to claim 9, wherein the method is carried out under an Argon inert atmosphere.

11. The method according to claim 4, wherein said first layer is a cobalt chrome alloy powder and said base or core is a cobalt chrome alloy.

12. The method according to claim 4, wherein said first layer is a titanium powder and said base or core is a cobalt chrome alloy.

13. The method according to claim 4, wherein said first layer is a niobium powder and said base or core is a cobalt chrome alloy.

14. The method according to claim 4, wherein said first layer is a tantalum powder and said base or core is a cobalt chrome alloy.

15. The method according to claim 4, wherein said first layer is a niobium powder and the base or core is a titanium alloy.

16. The method according to claim 4, wherein said first layer is a titanium powder and said base or core is a titanium alloy.

17. The method according to claim 4, wherein said first layer is a tantalum powder and said base or core is a titanium alloy.

5 18. The method according to claim 4, wherein said first layer is a stainless steel alloy and said base or core is a stainless steel alloy.

19. The method according to claim 4, wherein said first layer is a titanium powder and said base or core is a stainless steel alloy.

10 20. The method according to claim 4, wherein said first layer is a tantalum powder and said base or core is a stainless steel alloy.

15 21. The method according to claim 4, wherein said first layer is a niobium powder and said base or core is a stainless steel alloy.

22. The method according to claim 4, wherein said first layer is a niobium powder and said base or core is niobium.

20 23. The method according to claim 4, wherein said first layer is a tantalum powder and said base or core is tantalum.

25 24. The method according to claim 4, wherein said first layer is a niobium powder and said base or core is a cobalt chromium alloy, wherein said niobium forms a bond coat between said cobalt chromium alloy and said niobium powder, said successive layers of powder including at least one layer of titanium powder are deposited onto said bond coat forms the ingrowth structure.

30 25. The method according to claim 4, wherein said first layer is a tantalum powder and said base or core is a cobalt chromium alloy, wherein said tantalum powder forms a bond coat between said cobalt chromium alloy and said tantalum

powder, said successive layers of powder including at least one layer of titanium powder are deposited onto said bond coat which forms the ingrowth structure.

26. The method according to claim 4, wherein said  
5 first layer is tantalum powder and said base or core is a stainless steel alloy, wherein said tantalum powder forms a bond coat between said stainless steel alloy and said tantalum powder, said successive layers of powder including at least one  
10 layer of titanium powder are deposited onto said bond coat which forms the ingrowth structure.

27. The method according to claim 4, wherein said first layer is a niobium powder and said base or core consists of a stainless steel alloy, wherein said niobium powder forms a  
15 bond coat between said stainless steel alloy and said niobium powder, said successive layers of powder including at least one layer of titanium powder are deposited onto said bond coat which forms the ingrowth structure.

28. The method according to claim 4, wherein said core is integral with said resultant porous ingrowth structure  
20 and imparts additional physical properties to the overall construct.

29. The method accordingly to claim 4, wherein said core is detached from a resultant porous surface buildup.

30. The method according to claim 4, wherein a third  
25 element is added between said base and said first layer of powder to form a bond coat on said substrate.

31. A method of producing a three-dimensional porous tissue ingrowth structure comprising:

30 depositing a first layer of powder made from a metal selected from the group consisting of titanium, titanium alloys, stainless steel, cobalt chrome alloys, tantalum and niobium, onto a substrate;

scanning a laser beam having a power and a diameter with a speed between approximately  $100 \text{ mms}^{-1}$  to  $260 \text{ mms}^{-1}$  and a beam overlap between approximately +40% and -50% over said powder; and

5 depositing at least one additional layer of said powder onto said first layer and repeating said laser scanning steps for each successive layer.

32. The method according to claim 31, wherein the thickness of each of said layers is between  $50 \mu\text{m}$  -  $2000 \mu\text{m}$ .

10 33. The method according to claim 31, wherein said substrate is a base or core made of a metal selected from the group consisting of titanium, titanium alloys, stainless steel, cobalt chrome alloys, tantalum and niobium, wherein said first layer is fused to said base or core.

15 34. The method according to claim 33, wherein the thickness of said powder is approximately  $100 \mu\text{m}$ .

35. The method according to claim 31, wherein said laser power is approximately 78 W to 82 W.

20 36. The method according to claim 31, wherein said laser power is applied in a continuous wave.

37. The method according to claim 31, wherein said laser power is applied in a pulse wave.

25 38. The method according to claim 37, wherein said pulse wave repetition rate is between approximately 20 KHz to 50 KHz.

39. The method according to claim 31, wherein said method is carried out under an inert atmosphere.

40. The method according to claim 39, wherein the method is carried out under an Argon inert atmosphere.

30 41. The method according to claim 33, wherein said first layer is a cobalt chrome powder and said base or core is a titanium alloy.

42. The method according to claim 31, wherein said scanning speed is between  $180 \text{ mms}^{-1}$  and  $200 \text{ mms}^{-1}$ .

43. The method according to claim 33, wherein said first layer of powder is niobium and said base or core consists of a titanium alloy, wherein said laser beam is a 72 W to 92 W continuous wave with a scanning speed range between about  $160 \text{ mms}^{-1}$  to  $240 \text{ mms}^{-1}$  and said beam overlap is between 40% to -500%.

44. The method according to claim 43, wherein said scanning speed is between approximately  $190 \text{ mms}^{-1}$  and  $240 \text{ mms}^{-1}$ .

45. The method according to claim 43, wherein said beam overlap is 40%.

46. The method according to claim 33, wherein said first layer of powder is tantalum and said solid base or core is a titanium alloy, wherein said laser beam is a 72 W to 92 W continuous wave with a scanning speed range between about  $160 \text{ mms}^{-1}$  to  $240 \text{ mms}^{-1}$  and said beam overlap is between +40% to -500%.

47. The method according to claim 46, wherein said scanning speed is between about  $160 \text{ mms}^{-1}$  to  $200 \text{ mms}^{-1}$ .

48. The method according to claim 46, wherein said beam overlap is -40% and the laser beam power is 82 W.

49. The method according to claim 33, wherein said first layer of powder is titanium and said solid base or core is a cobalt chrome alloy, wherein said laser beam is a 72 W to 92 W continuous wave with a scanning speed range between about  $160 \text{ mms}^{-1}$  to  $240 \text{ mms}^{-1}$  and said beam overlap is between +40% to -500%.

50. The method according to claim 49, wherein said scanning speed range is between about  $160 \text{ mms}^{-1}$  to  $200 \text{ mms}^{-1}$ .

51. The method according to claim 49, wherein said beam overlap is -40% and said laser beam has a power of 82 W.

52. The method according to claim 33, wherein said first layer of powder is titanium and said solid base or core is a titanium alloy, wherein said laser beam is a 72 W to 92 W continuous wave with a scanning speed range between about 160 mms<sup>-1</sup> to 240 mms<sup>-1</sup> and said beam overlap is between about +40% to -500%.

53. The method according to claim 52, wherein said scanning speed is between 160 mms<sup>-1</sup> to 200 mms<sup>-1</sup>.

54. The method according to claim 52, wherein said beam overlap is -40% and said laser beam has a power of 82 W.

55. The method according to claim 33, comprising the additional step of subjecting a powder layer to a second laser scan with a scanning speed and beam overlap in an orthogonal direction to a first scan.

56. A method according to claim 55, wherein said scanning speed is between 160 mms<sup>-1</sup> to 200 mms<sup>-1</sup>.

57. A method according to claim 55, wherein said beam overlap is between -400% to -600%.

58. The method according to claim 31, comprising the additional step of subjecting a powder layer to a second laser scan with a scanning speed and beam overlap in an orthogonal direction to a first scan.

59. A method according to claim 58, wherein said scanning speed is between 160 mms<sup>-1</sup> to 200 mms<sup>-1</sup>.

60. A method according to claim 58, wherein said beam overlap is between -400% to -600%.

DATED: 6 NOVEMBER 2003

PHILLIPS ORMONDE & FITZPATRICK

ATTORNEYS FOR:

HOWMEDICA OSTEONICS CORP.

*David B Fitzpatrick*

1/28

FIG. 1

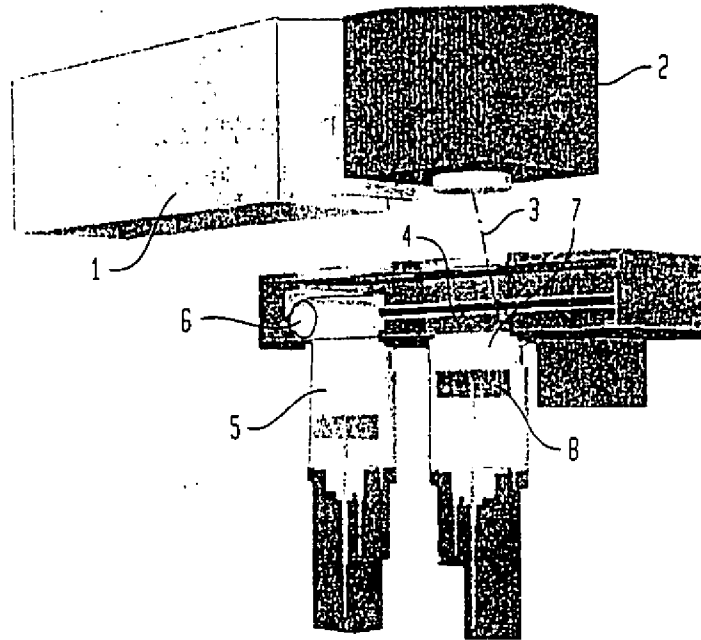
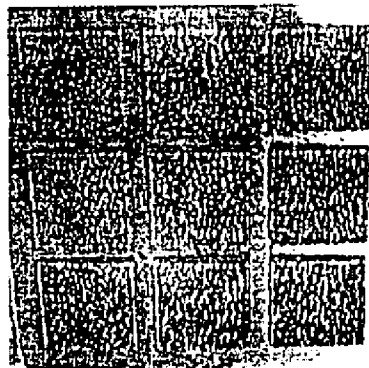


FIG. 2



2/28  
**FIG. 3**

METAL POWDER	MEAM OVERLAP (%)	-500										-250							
	LASER SCANNING SPEED (mm/s)	100	120	140	160	180	200	220	240	260	100	120	140	160	180	200	220	240	
	SUBSTRATE																		
CoCr	Ti ALLOY																		
	CoCr																		
	STAINLESS STEEL	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	
Nb	Ti ALLOY																		
	CoCr																		
	STAINLESS STEEL																		
Ta	Ti ALLOY																		
	CoCr																		
Ti	Ti ALLOY																		
	CoCr																		

O	MEAM OVERLAP (%)	-40										25							
	LASER SCANNING SPEED (mm/s)	160	170	180	190	200	210	220	230	240	100	120	140	160	180	200	220	240	
	SUBSTRATE																		
CoCr	Ti ALLOY										✓	✓	✓	✓	✓	✓	✓	✓	
	CoCr										✓	✓	✓	✓	✓	✓	✓	✓	
	STAINLESS STEEL										✓	✓	✓	✓	✓	✓	✓	✓	
Nb	Ti ALLOY	✓	✓	✓	✓	✓	✓	✓	✓	✓									
	CoCr	✓	✓	✓	✓	✓	✓	✓	✓	✓									
	STAINLESS STEEL	✓	✓	✓	✓	✓	✓	✓	✓	✓									
Ta	Ti ALLOY	✓	✓	✓	✓	✓	✓	✓	✓	✓									
	CoCr	✓	✓	✓	✓	✓	✓	✓	✓	✓									
Ti	Ti ALLOY	✓	✓	✓	✓	✓	✓	✓	✓	✓									
	CoCr	✓	✓	✓	✓	✓	✓	✓	✓	✓									

O	MEAM OVERLAP (%)	50										50							
	LASER SCANNING SPEED (mm/s)	100	120	140	160	180	200	220	240	260	280	300	320	340	360	380	400	420	
	SUBSTRATE																		
CoCr	Ti ALLOY																		
	CoCr																		
	STAINLESS STEEL	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	✓	
Nb	Ti ALLOY																		
	CoCr																		
	STAINLESS STEEL	✓	✓	✓	✓	✓	✓	✓	✓	✓									
Ta	Ti ALLOY																		
	CoCr																		
Ti	Ti ALLOY																		
	CoCr																		



FIG. 4A

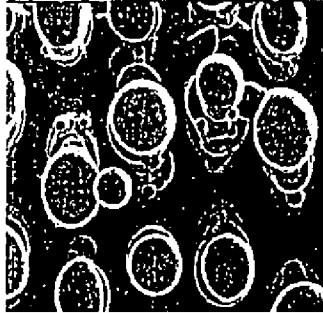


FIG. 4B

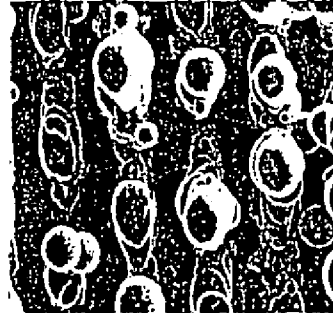


FIG. 4C

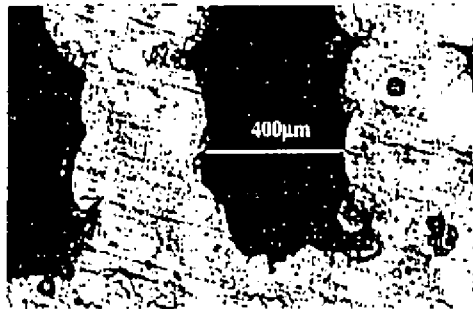


FIG. 5A



FIG. 5B



FIG. 6A

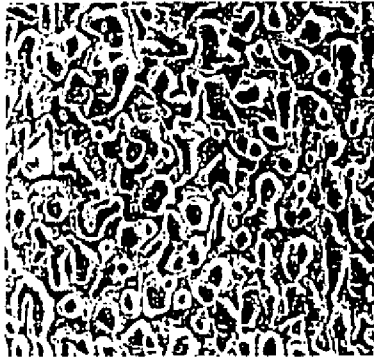


FIG. 6B

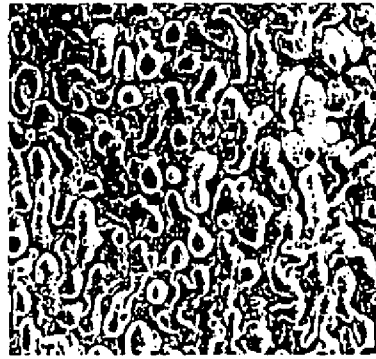


FIG. 6C

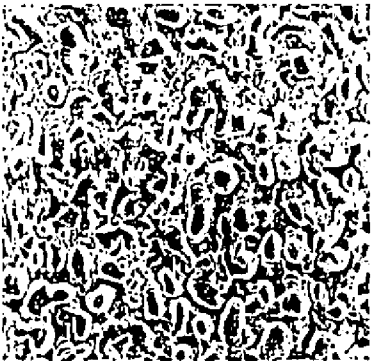


FIG. 6D

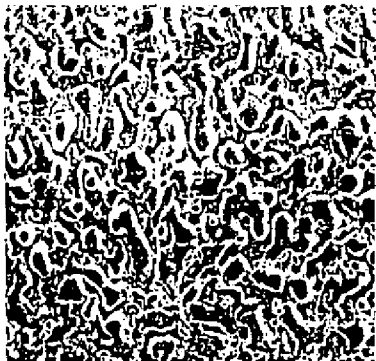


FIG. 6E

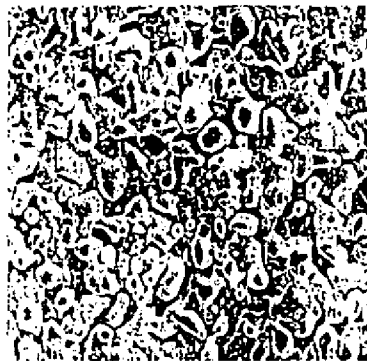


FIG. 7A



FIG. 7B

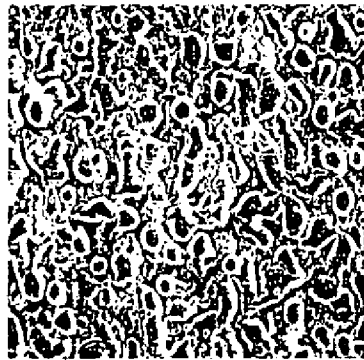


FIG. 7C

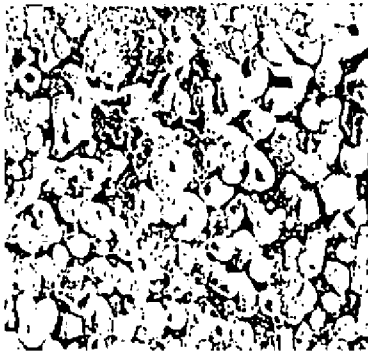


FIG. 8A



FIG. 8B

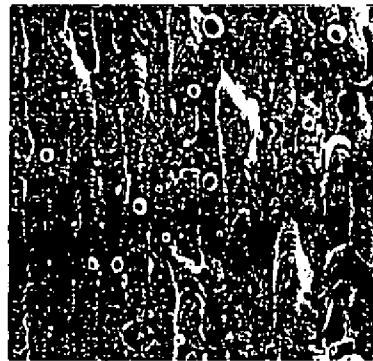


FIG. 8C



FIG. 8D



FIG. 8E

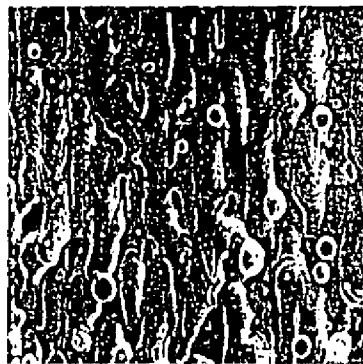


FIG. 9A



FIG. 9B

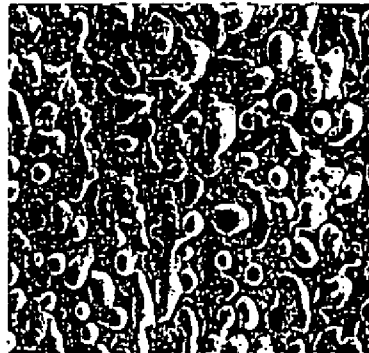


FIG. 9C

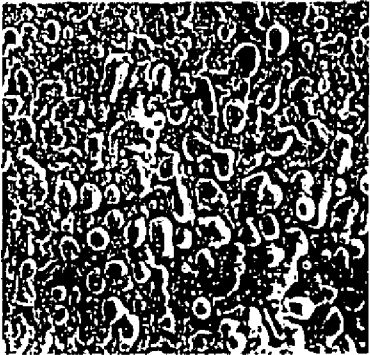


FIG. 9D

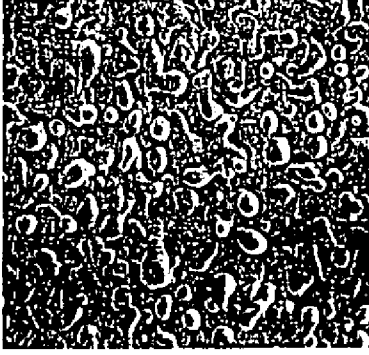


FIG. 9E

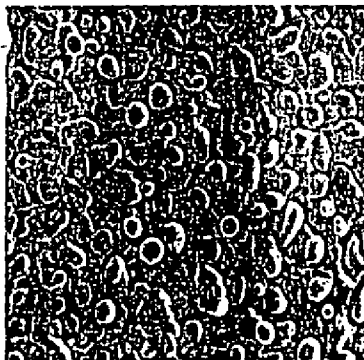


FIG. 10A



FIG. 10B

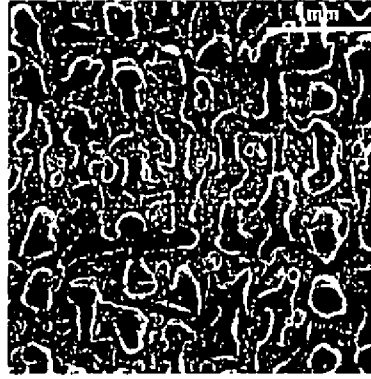


FIG. 10C



FIG. 10D

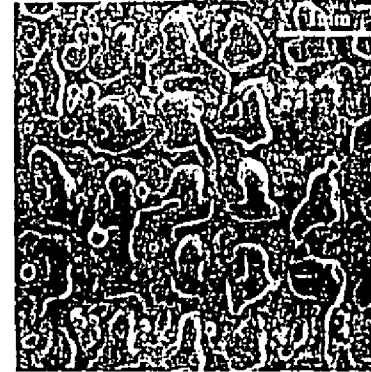


FIG. 10E

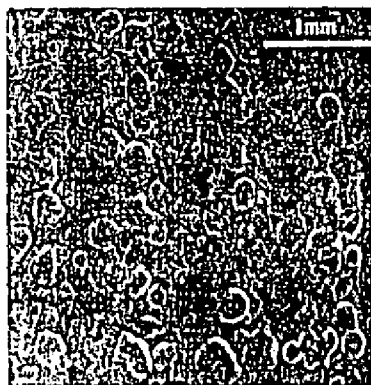


FIG. 10F

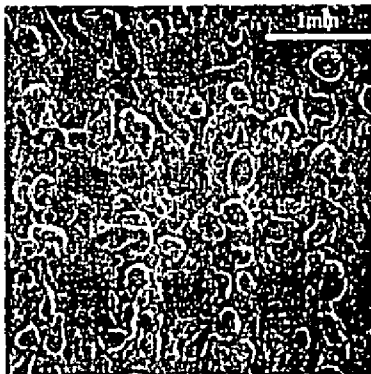
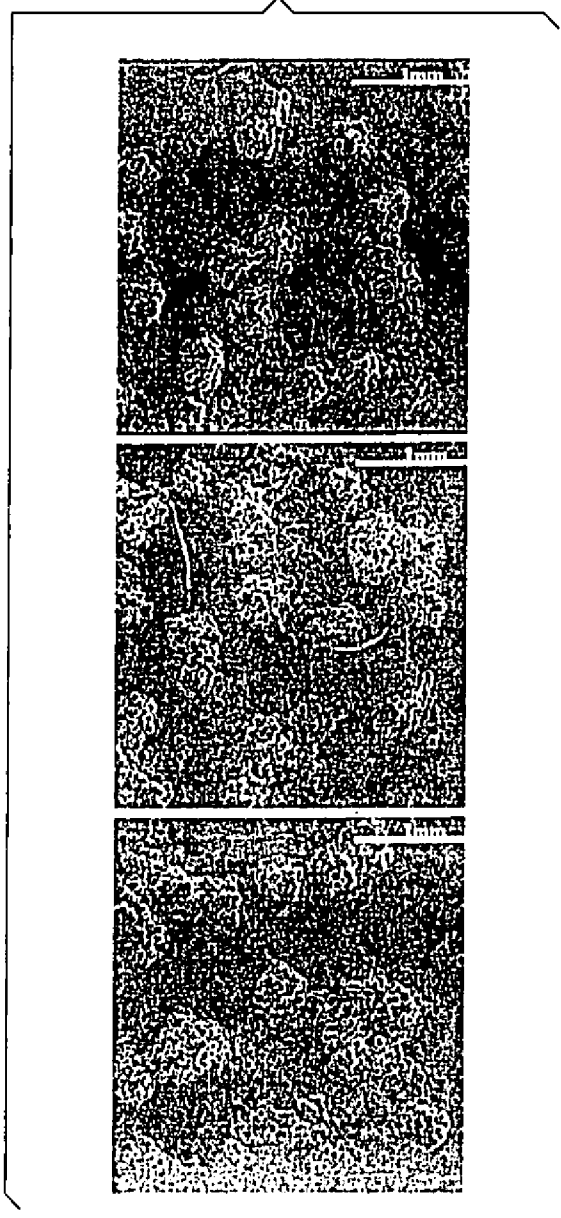


FIG. 11



10/28

FIG. 12

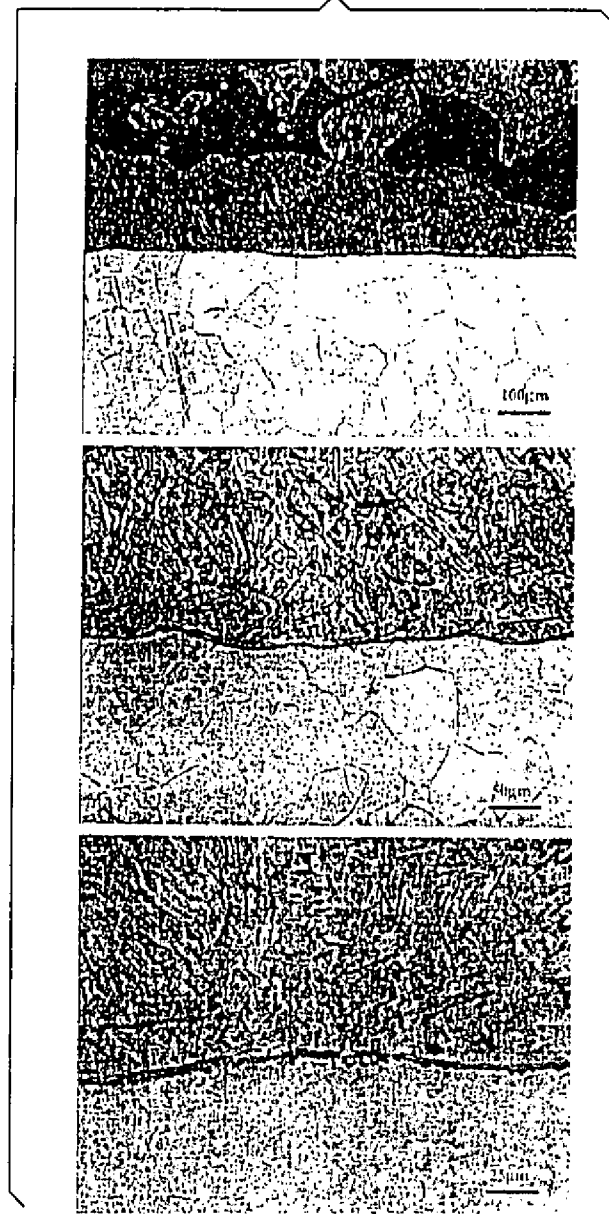




FIG. 13

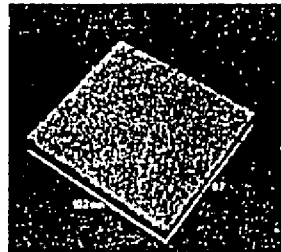
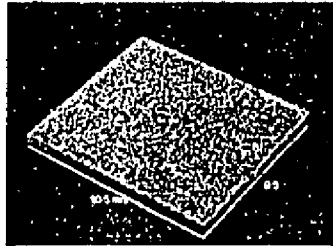
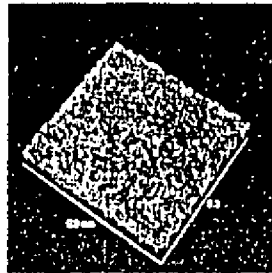
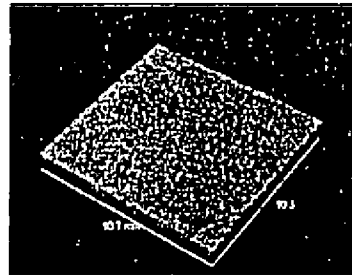


FIG. 14A

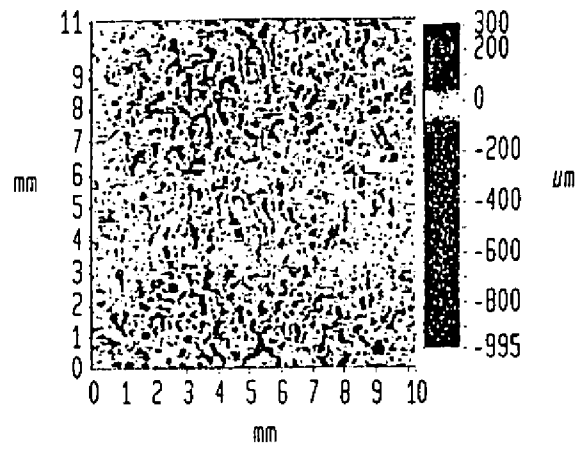


FIG. 14B

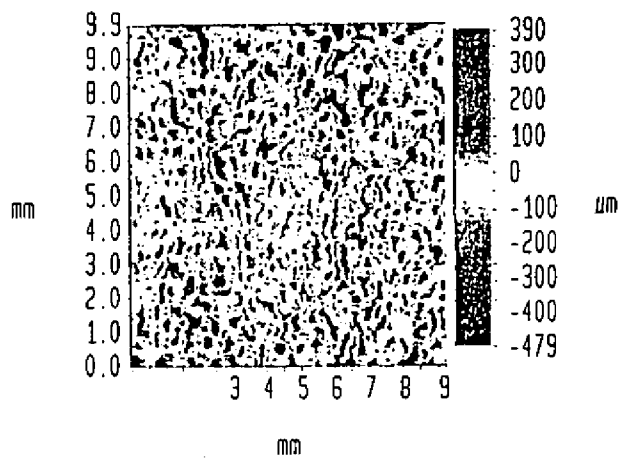


FIG. 15A

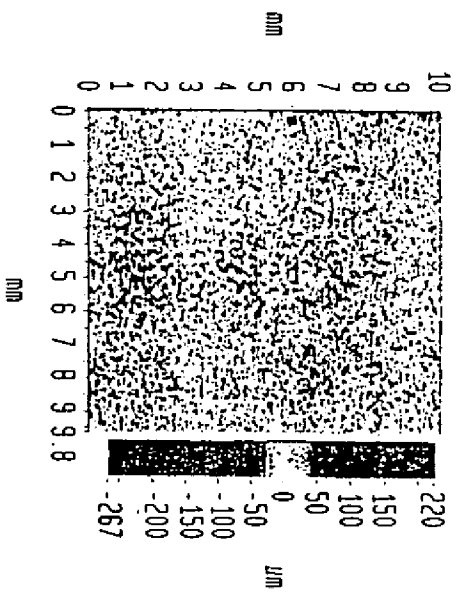


FIG. 15B

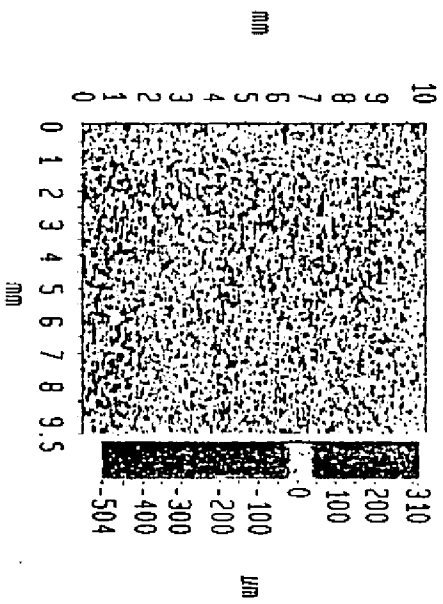


FIG. 16A

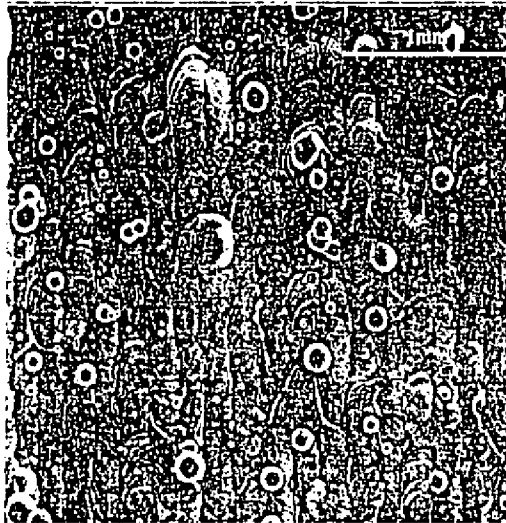


FIG. 16B

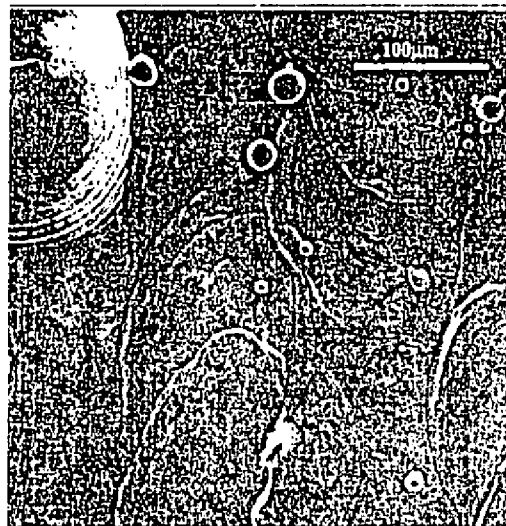


FIG. 17A

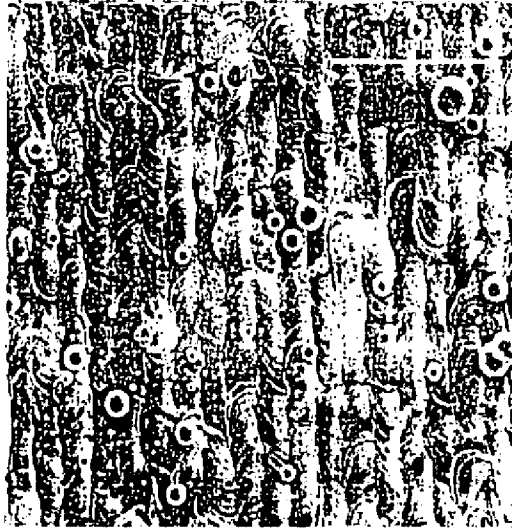


FIG. 17B

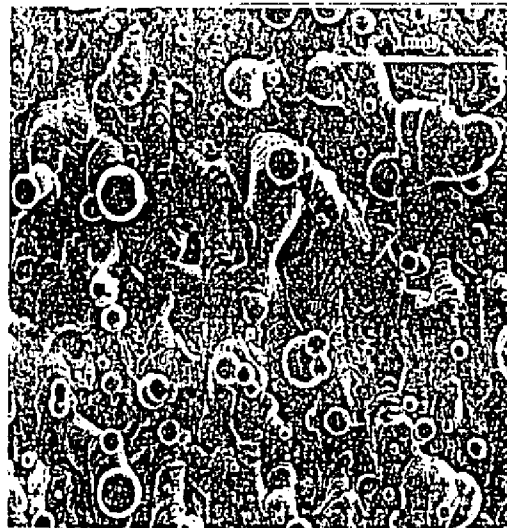


FIG. 18A

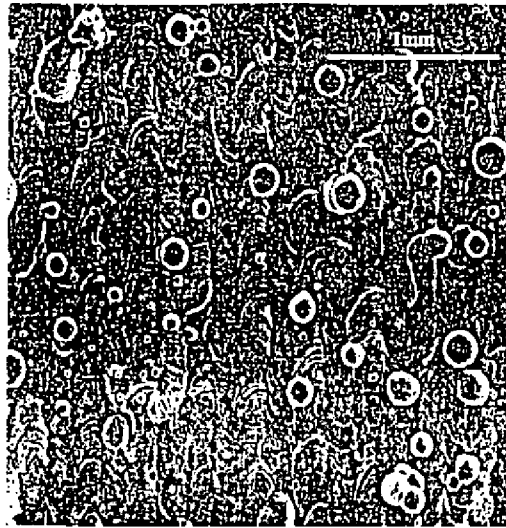


FIG. 18B

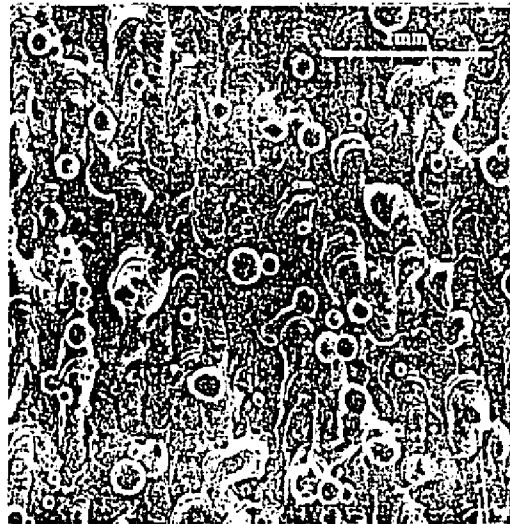


FIG. 19A

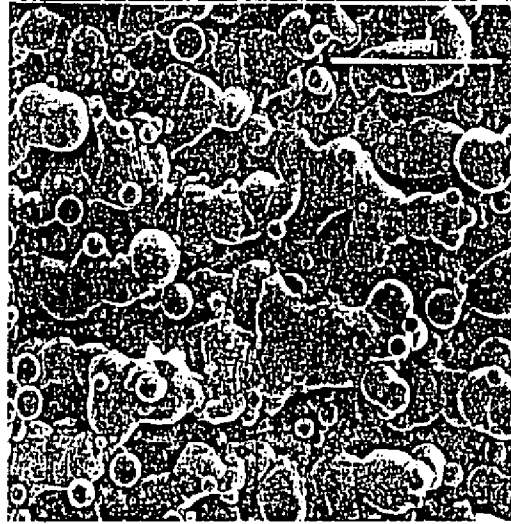


FIG. 19B



FIG. 20A

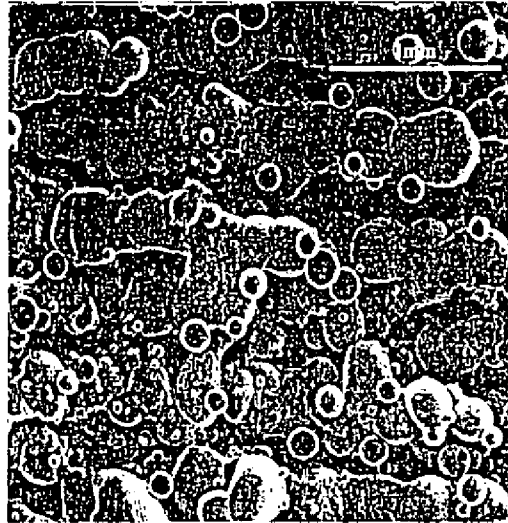


FIG. 20B

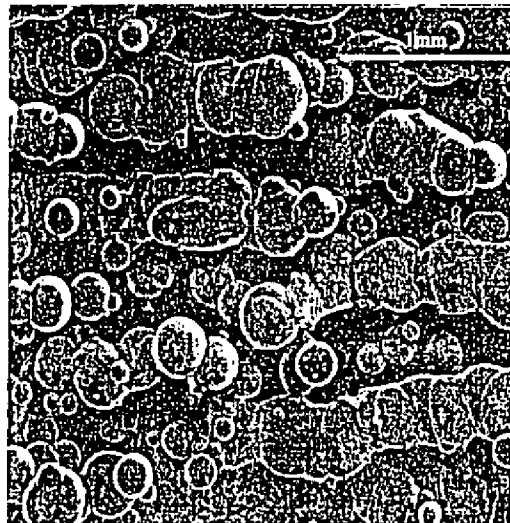




FIG. 21A

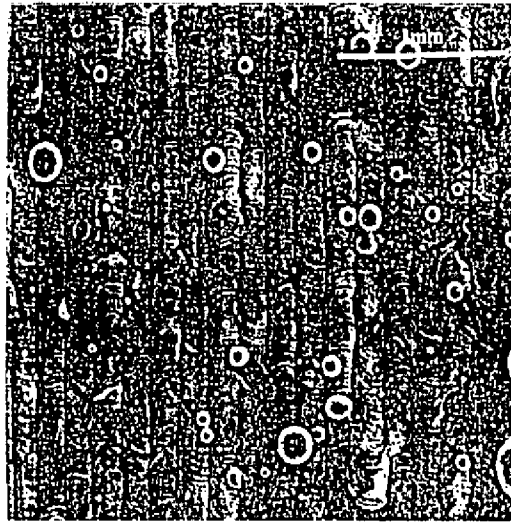


FIG. 21B

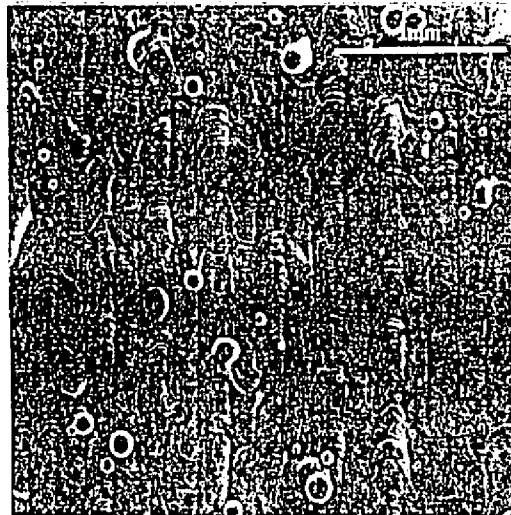


FIG. 22A

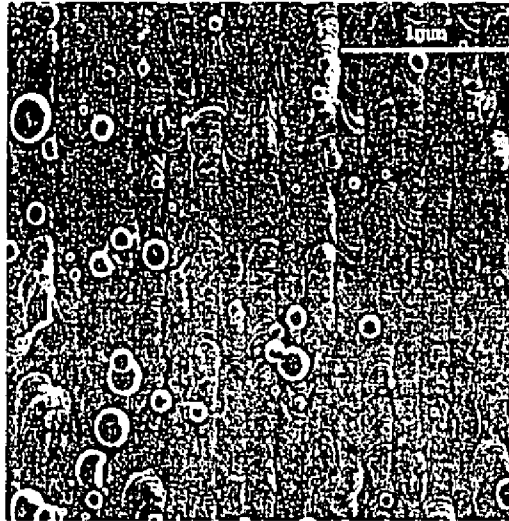


FIG. 22B

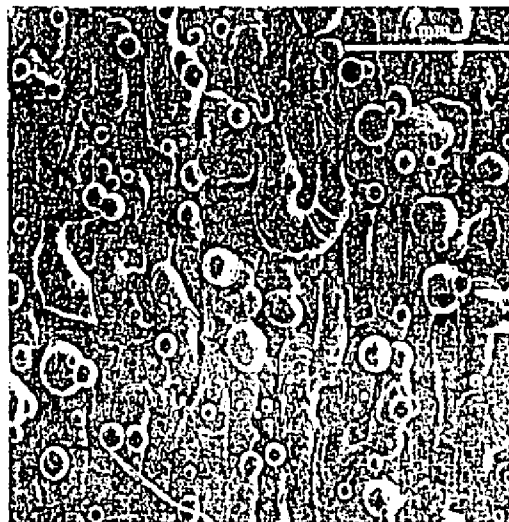


FIG. 23A

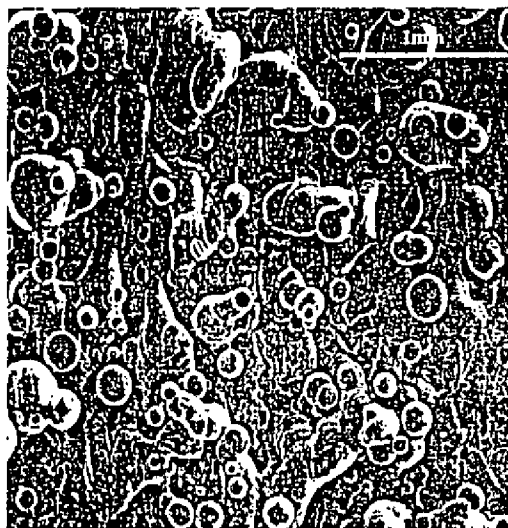


FIG. 23B

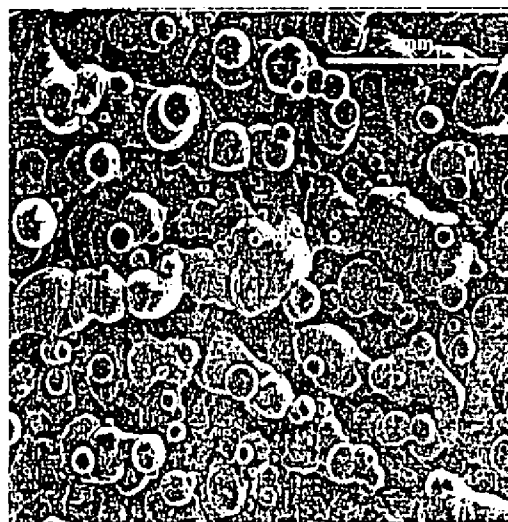


FIG. 24A

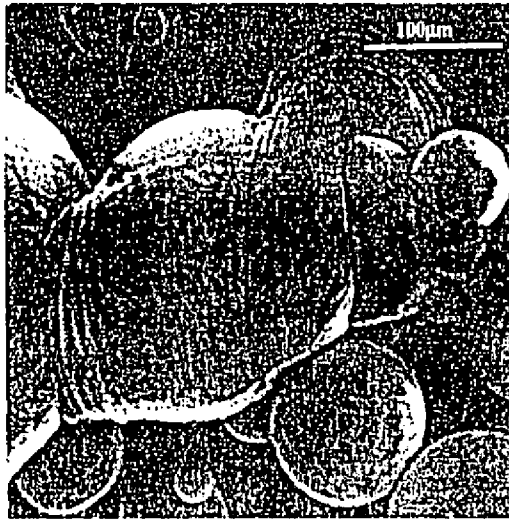


FIG. 24B

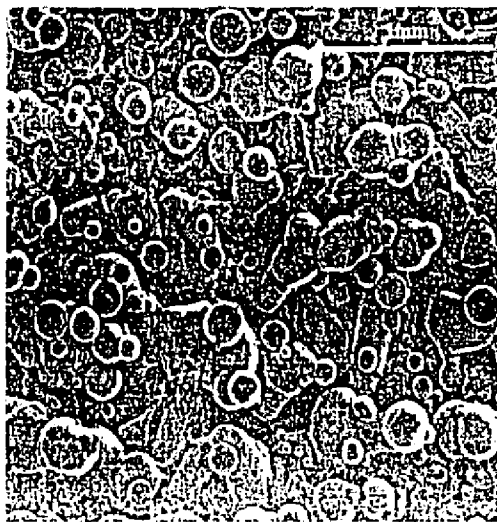


FIG. 25

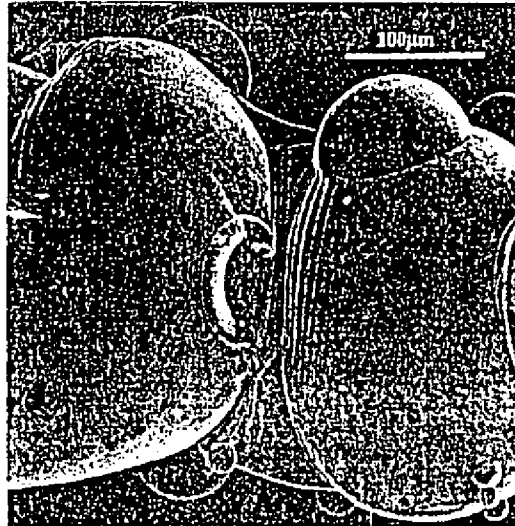


FIG. 26

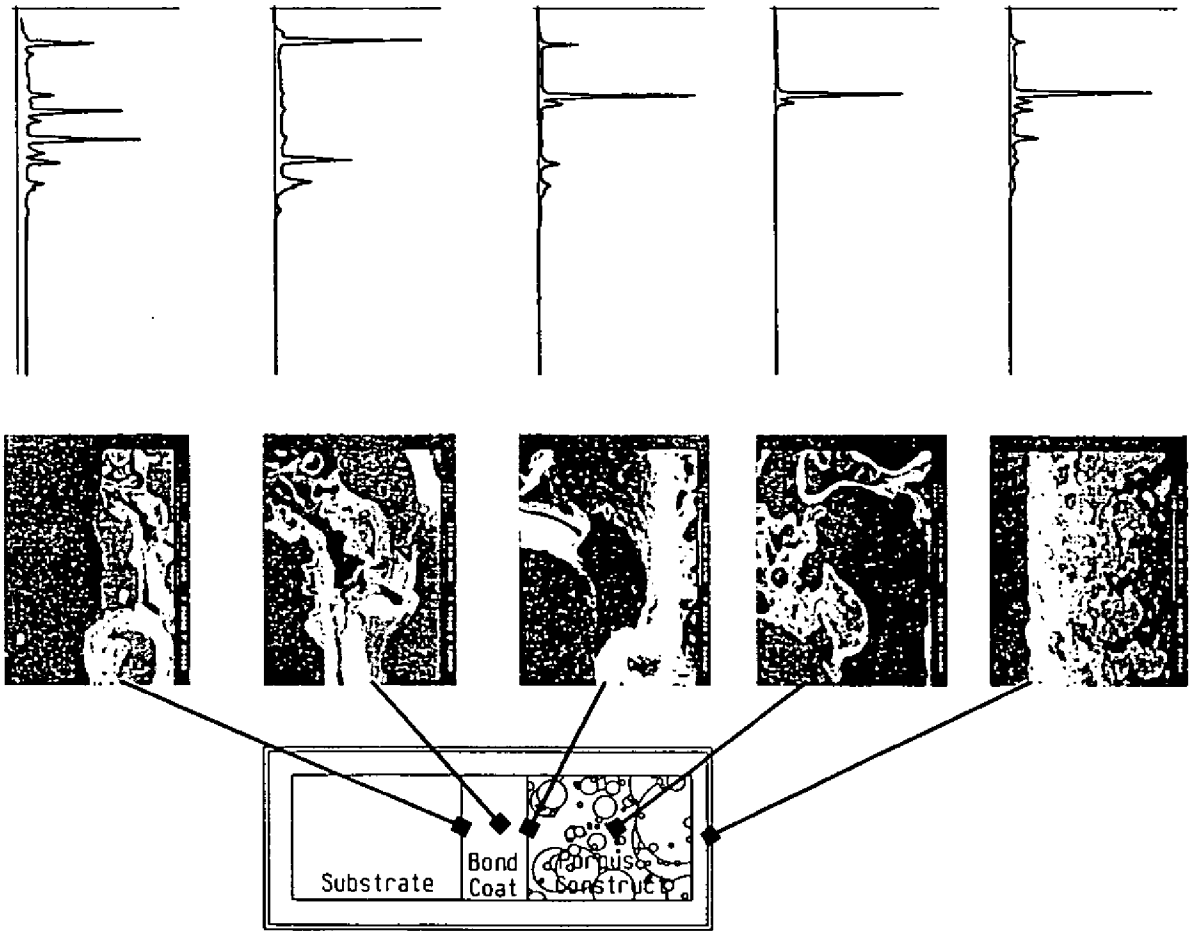


FIG. 27

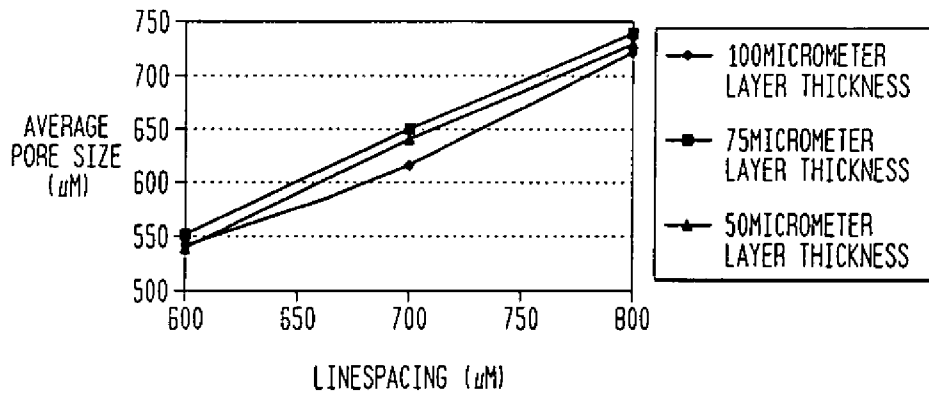


FIG. 28A

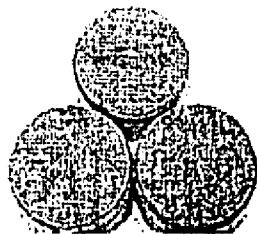


FIG. 28B

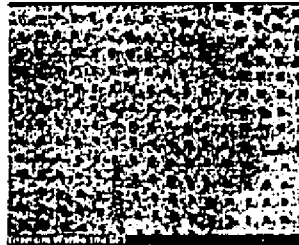


FIG. 28C

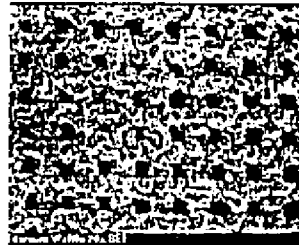


FIG. 28D



FIG. 28E



FIG. 28F



26/28

FIG. 29





FIG. 30

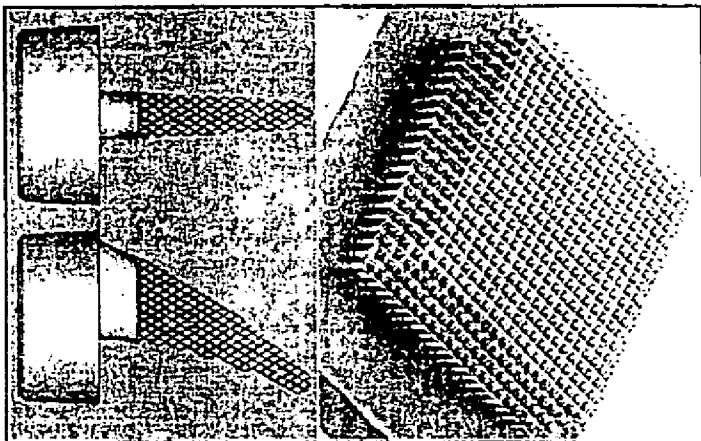


FIG. 31

